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for

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## CE

B.Tech

# STEEL

## Design of Steel Structure

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# CHAPTER - 1

## (GENERAL CONSIDERATION)

### Steel as a structural material:-

1. Steel members have high strength to weight ratio therefore a steel member of small sections which has little self weight as it is able to resist heavy load.
2. This property is most important for the construction of long span bridges, tall buildings and for buildings on soil with relatively low bearing capacity.
3. Steel is a ductile material does not fail suddenly but gives proper warning before failure.
4. Steel structures can be constructed very fast.
5. Steel structures can be easily dismantled and sold as scrap material.

### Disadvantages:-

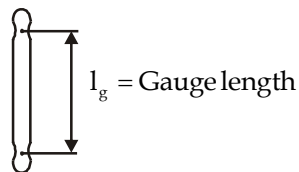
1. Steel structures when placed in exposed condition are subjected to corrosion therefore they required frequent maintenance.
2. Fire resistance technique is also required for steel structures.

### Structural steel (IS 2062):-

Grade	Ultimate tensile strength	Yield strength
Fe 410	410 N/mm <sup>2</sup>	250 N/mm <sup>2</sup>

Minimum % elongation	Gauge length for tensile test
23%	$= 5.65 \sqrt{A_0}$

$A_0$  = Initial cross-section area



Modulus of elasticity =  $2 \times 10^5$  N/mm<sup>2</sup>

Shear modulus =  $0.769 \times 10^5$  N/mm<sup>2</sup>

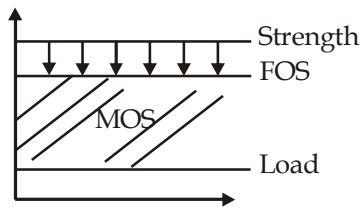
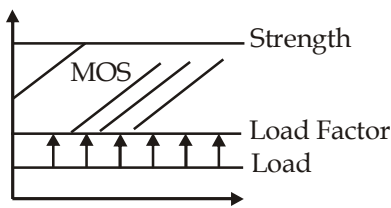
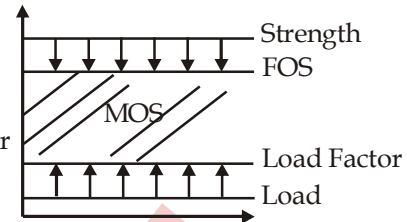
Poisson's ratio

Elastic Limit = 0.3

Plastic Limit = 0.5

Coefficient of thermal expansion =  $12 \times 10^{-6}/^\circ\text{C}$

Unit mass (G) =  $7.85 \times 10^3$  kg/m<sup>3</sup>

**Design philosophies:-****WSM****USM****LSM**

→ The uncertainties affecting the safety of a structure

1. The uncertainty about loading
2. Uncertainty about material strength
3. Uncertainty about structural dimension and behaviour

These uncertainty together makes it impossible to designer to guarantee that a structure will be absolutely safe. There is always a small risk of failure.

**WSM (IS 800 - 1984) - or elastic method.**

This method is based upon elastic theory attainment of yield stress forms the design criteria for the member.

The working stress should be less than permissible stress.

$$\text{Permissible stress} = \frac{\text{Yield stress}}{\text{FOS}}$$

Working stress (during design life)  $\leq$  Permissible stress

**Permissible stress in structural member:-**

Type of stress	Notations	Permissible stress	FOS
(i) Axial Tension	$\sigma_{at}$	$0.6 f_y$ $= 0.6 \times 250$ $= 150 \text{ N/mm}^2$	$0.6 f_y = \frac{f_y}{\text{FOS}}$ FOS = 1.67
(ii) Axial Compression	$\sigma_{ac}$	$0.6 f_y$	1.67
(iii) Bending Tension	$\sigma_{bt}$	$0.66 f_y$	1.5
(iv) Bending Compression	$\sigma_{bc}$	$0.66 f_y$	1.5
(v) Average shear	$\tau_{va}$	$0.4 f_y$	2.5
(vi) Maximum shear	$\tau_{vm}$	$0.45 f_y$	2.22
(vii) Bearing stress	$\sigma_p$	$0.75 f_y$	1.33
(viii) Stress in slab base	$\sigma_{bs}$	$185 \text{ N/mm}^2$	—

One of the major drawback of designing steel structure by WSM approach the reserved strength beyond elastic limit is not utilized.

**Limit state method:-** The limit state design method was developed to take account of all conditions that can make the structure unfit for use. This theory is based upon a probabilistic approach.

**Limit state of collapse or strength :-** Limit state of strength is considered that the structure should be safe under the action of most unfavorable load combinations.

**Limit state serviceability:-** Structure should be serviceable at working load conditions. Examples of serviceability criteria are deflection, vibrations, corrosion, fire resistance.

**Characteristic strength (Su):-** Characteristic strength of a material is defined as the value of resistance below which not more than 5% test results are expected to fail.

$$\text{Design strength} = \frac{f_u \text{ (characteristic strength)}}{\text{FOS}}$$

**Characteristic load (Qu):-** Characteristic load is a load which has 95% probability of not being exceeded during the design life of the structure.

$$\text{Design load} = \text{Load Factor} \times Q_u$$

$$\begin{aligned} \text{Failure Probability} &= 0.05 \times 0.95 + 0.05 \times 0.95 + 0.05 \times 0.05 \\ &= 0.0975 = 9.75\% \approx 10\% \end{aligned}$$

#### Partial safety factors for materials

	FOS
1. Resistance governed by yielding	1.10
2. Resistance governed by buckling	1.10
3. Resistance governed by ultimate strength	1.25

#### Partial safety factors for connections

	Shop Fabric	Field Fabric
1. Bolted connections	1.25	1.25
2. Welded connections	1.25	1.50

# CHAPTER - 2

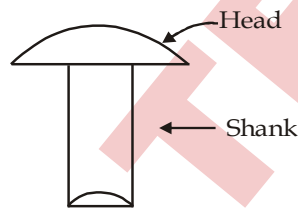
## (CONNECTION DESIGN - RIVETED CONNECTION)

### Connection Design:-

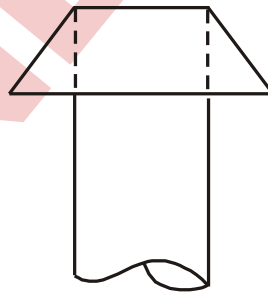
1. Riveted Connections (WSM)
2. Bolted Connections (LSM)
3. Welded Connections (WSM and LSM)

### Riveted Connections:-

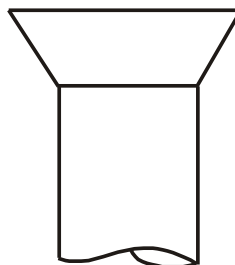
(i) **Snap Head Rivet:-**



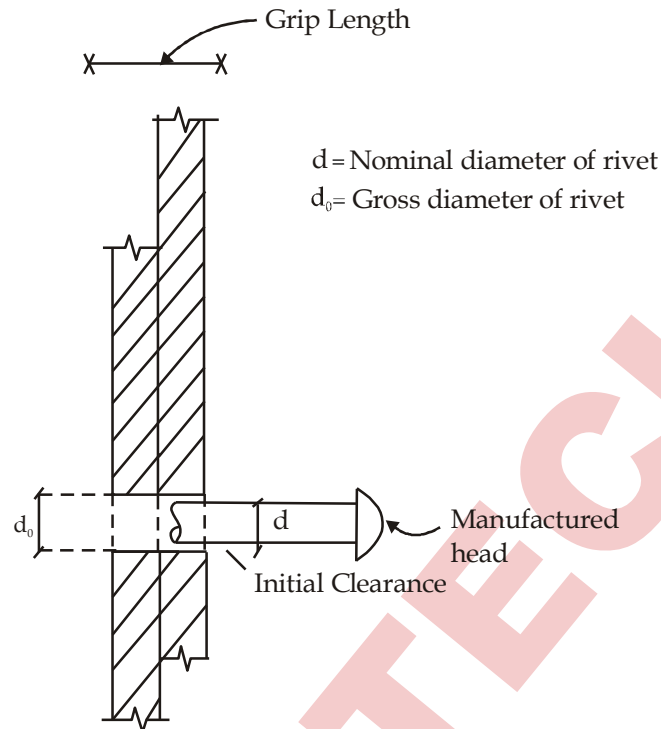
(ii) **Pan Headed Rivet:-**



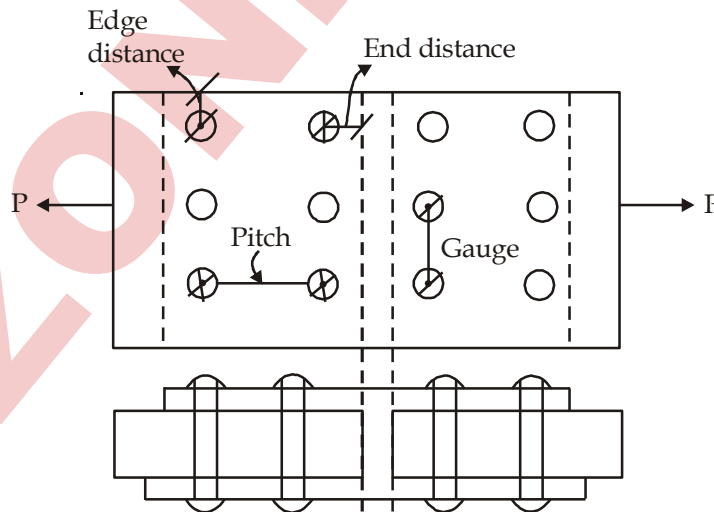
(iii) **Counter Shank Rivet:-**







### Some important specifications:-



(i) Diameter of rivet hole =  $d_0$ :-

$$d_0 = d + 1.5 \text{ mm (} d \leq 25 \text{ mm)}$$

$$d_0 = d + 2 \text{ mm (} d > 25 \text{ mm)}$$

2. **Minimum Pitch (p):-** Minimum pitch is ensure for the following reasons:-
- To prevent bearing failure of members between two rivets.
  - To permit efficient intallation of rivets to ensure sufficient space for riveting.

Minimum pitch =  $2.5 \times$  nominal of rivet.

$$P_{\text{minimum}} = 2.5d$$

3. Minimum end or edge distance:-

=  $1.5d_o$  (For machine cut elements)

=  $1.7d_o$  (For hand cut elements)

4. Maximum edge or end distance:-

$$= 12 t_{\epsilon}$$

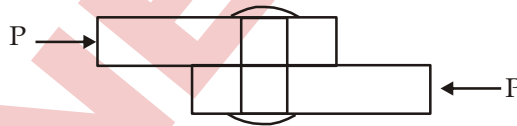
$$\epsilon = \sqrt{\frac{250}{f_y}}$$

$\epsilon$  = Yield ratio

t = Thickness of thinner member of connection

$f_y$  = yield strength = 250 N/mm<sup>2</sup>

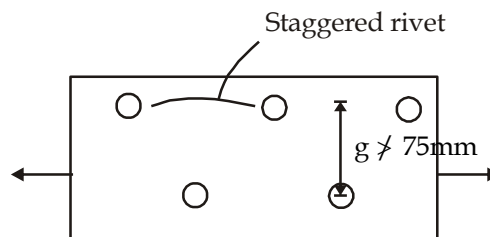
5. Maximum pitch (p):-



$$P_{\text{maximum}} = \begin{cases} 12t \text{ or } 200 \text{ mm whichever is minimum, Compression} \\ 16t \text{ or } 200 \text{ mm whichever is minimum, Tension} \end{cases}$$

t = Thickness of thinner member connected

In case of staggered rivet these values should be multiply by 1.5 times

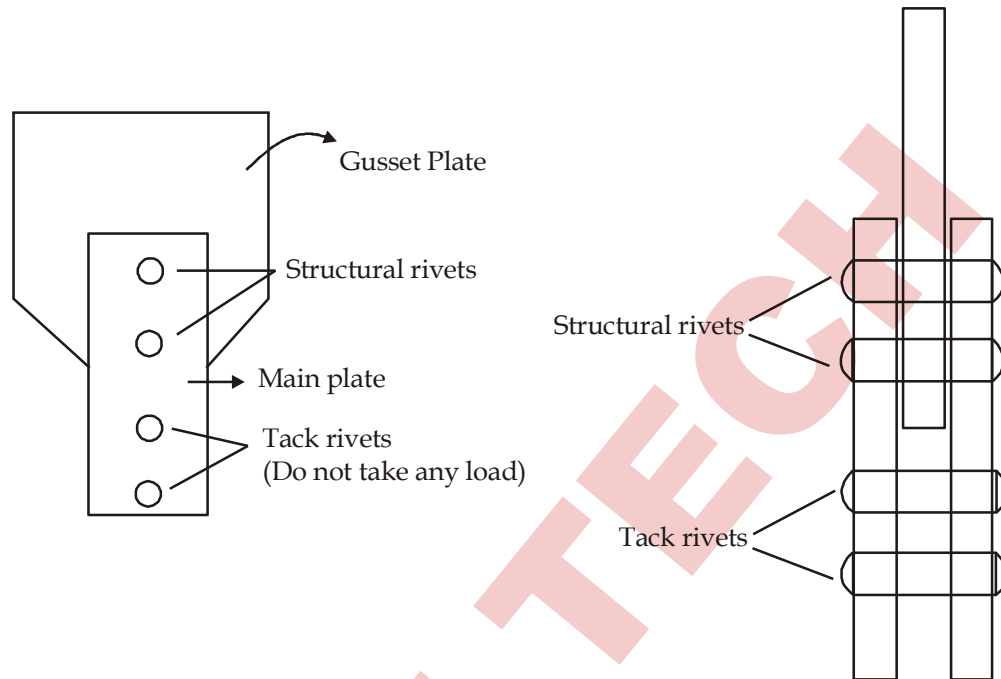


Compression (p)  $\neq$  [18t or 300 mm] min

Tension (p)  $\neq$  [24t or 300 mm] min

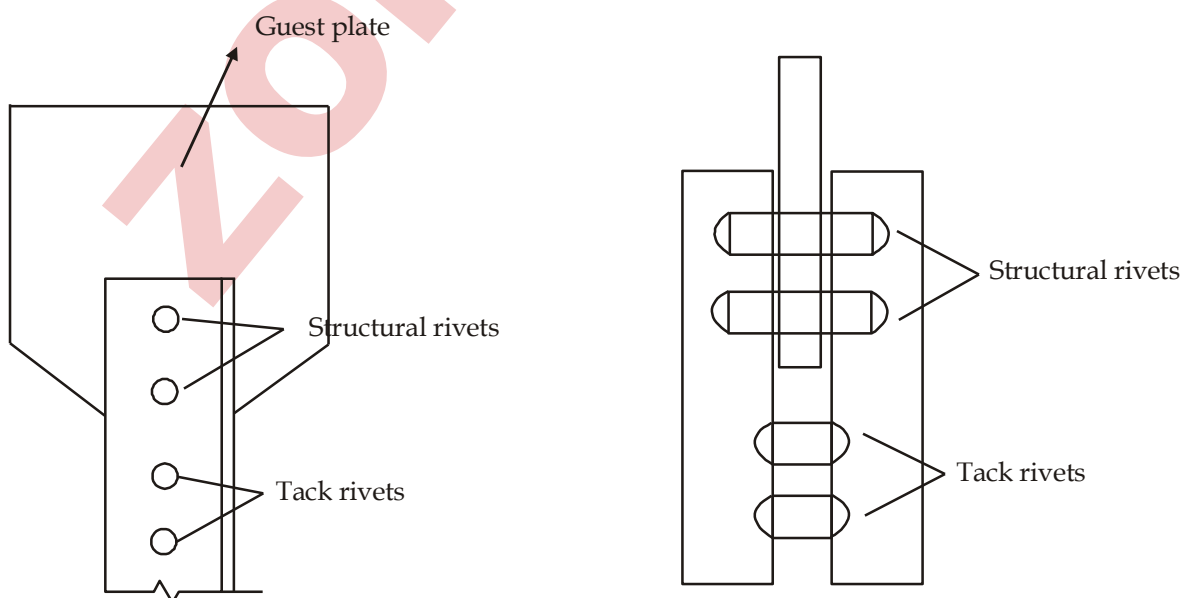
**Tack riveted:-**

1. When plate elements are tack riveted:-



Maximum  $p \neq [32t \text{ or } 300 \text{ mm}] \text{ min}$

2. When angle sections or channel sections are tack riveted:-





If Compression (P)  $\neq$  600 mm

Tension (P)  $\neq$  1000 mm

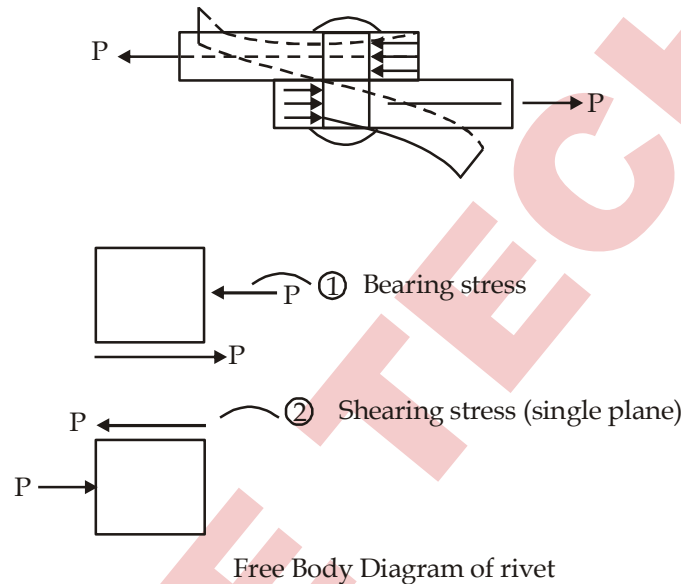
- To find nominal dia of rivet unwin's formula is used.

$$d = 6.05\sqrt{t} = \text{mm}$$

t = thickness of thinner member connected

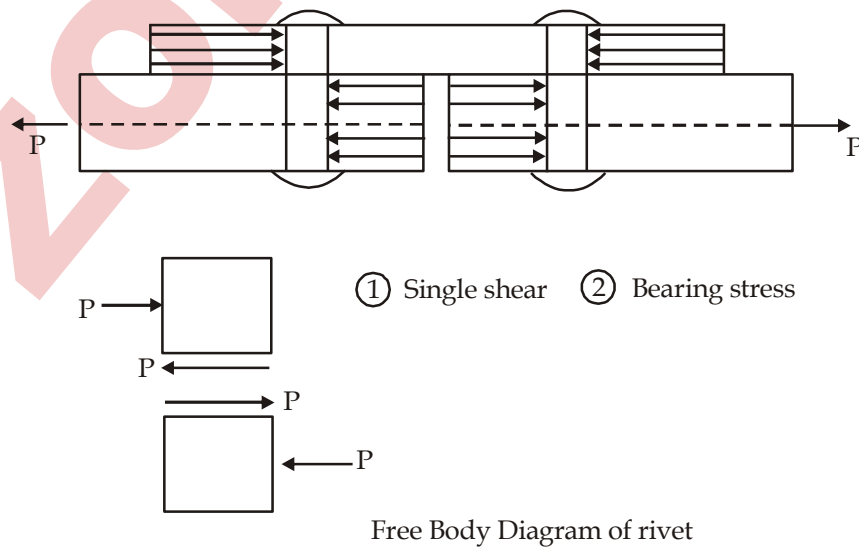
**Types of joints:-**

1. **Lap Joint:-**



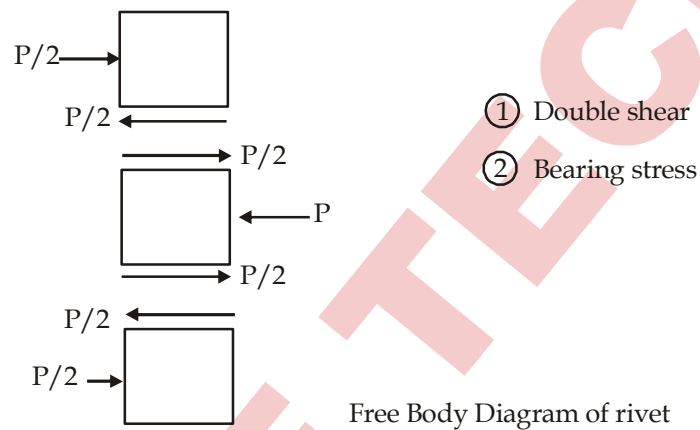
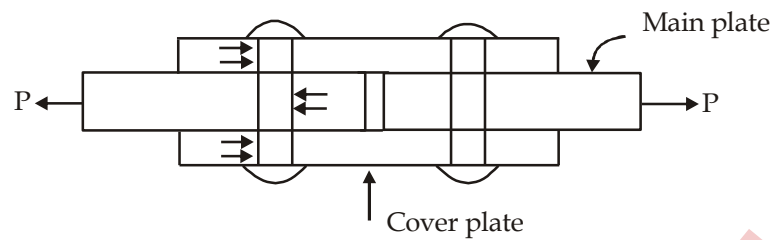
- Less efficient joint because line of action of forces are not same due to this bending stresses are generated in rivet.

2. **Single Cover Butt Joint:-**

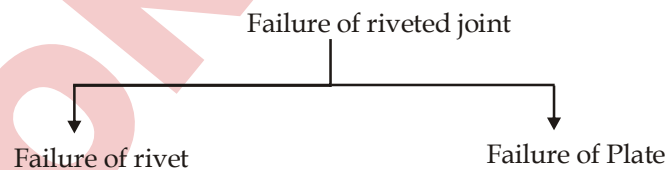


Line of action of load is same but connection is not symmetric about load line so slightly more efficient than lap joint.

## 2. Double Cover Butt Joint:-



Most efficient joint because load line is same and connection is also symmetric about load line.

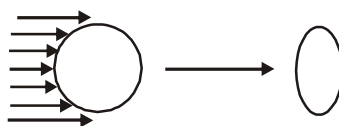


### Failure of rivet:-

- (1) Shear failure of rivet:-

Rivet get cut into two pieces

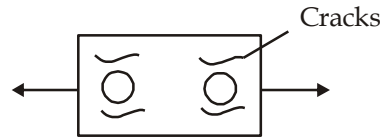
- (2) Bending failure of rivet



Circular c/s changed to elliptical

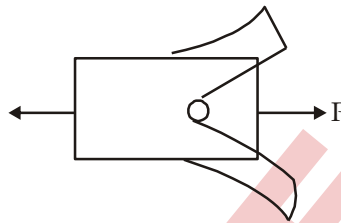
**Failure of Plate:-**

- (1) Shear failure of plate



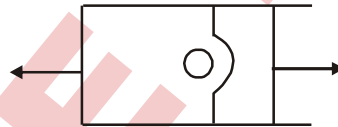
Cracks generated in the direction of force.

- (2) Splitting failure of plate



Due to diagonal tension

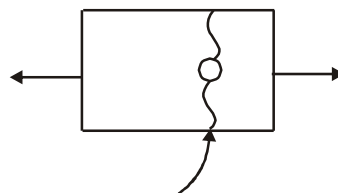
- (3) Bearing failure of plate



When rivet material is strong in comparison to plate material.

Rivet will push the plate forward.

- (4) Tearing failure or tension failure of plate



Cracks are generated in the perpendicular direction of load.

**Note:-**

1. By proper end or edge distance we can prevent shear failure, splitting failure and bearing failure of plate.
2. For a efficient design we have to ensure that shearing and bearing strength of rivet is more than tearing strength of plate (by it we can fully utilize the strength of plate (structural member)).

**Strength of riveted joint:-**

$$\text{Minimum of } \begin{cases} \text{Shearing strength of all rivets (P}_s\text{)} \\ \text{Bearing strength of all rivets (P}_b\text{)} \\ \text{Tearing strength of all rivets (P}_T\text{)} \end{cases}$$

**Permissible stresses in rivet:-**

	Permissible shear stress ( $\tau_{vf}$ )	Permissible bearing stress ( $\sigma_{pf}$ )
Shop rivet	Power driven	300 N/mm <sup>2</sup>
	Hand driven	250 N/mm <sup>2</sup>
	80 N/mm <sup>2</sup>	

**Note:-**

In case of field riveting above values are reduced by 10%.

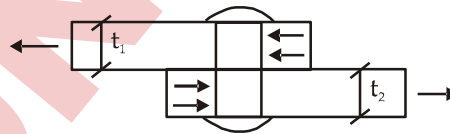
**1. Strength of lap joint:-****(a) Shearing strength of single rivet:-**

$$P_s = \frac{\pi}{4} d_o^2 \times \tau_{vf}$$

$d_o$  = dia of hole or gross dia of hole

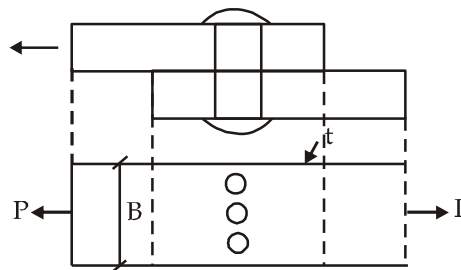
**(b) Bearing strength of single rivet:-**

$$P_b = d_o \cdot t \times \sigma_{pf}$$



$t$  = Thickness of thinner plate

$$\min \begin{cases} t_1 \\ t_2 \end{cases}$$

**(c) Tearing strength of plate:-**

$P_T$  = Tearing strength of plate

$$P_T = (B - nd_o) \times t \times \sigma_{at}$$

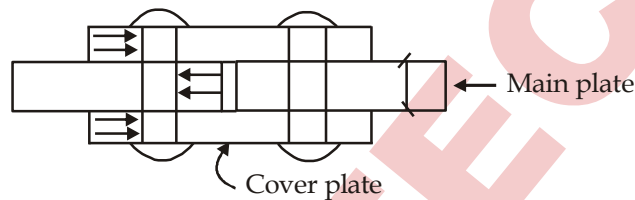
$$\text{Rivet Value } (R_V) = \min \begin{bmatrix} P_s \\ P_b \end{bmatrix}$$

$$\text{Number of rivet} = \frac{F \text{ (Load to be resisted)}}{R_V}$$

## 2. Double Cover Butt Joint:-

### (a) Shearing strength of single rivet:-

$$P_s = 2 \times \frac{\pi}{4} \cdot d_o^2 \cdot \tau_{vf}$$

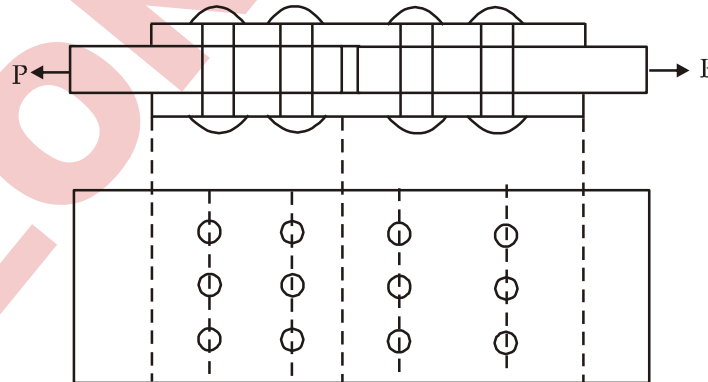


### (b) Bearing strength of single rivet:-

$$P_b = d_o \times t \times \sigma_{pf}$$

$$t = \min \begin{bmatrix} \text{summation of thickness of cover plates} \\ \text{Thickness of main plate} \end{bmatrix}$$

### Strength of all rivets



$$\text{Strength of all rivets} = n \times R_V$$

$$n = \text{Number of rivets} = 6 \text{ (not 12)}$$

### (c) Tearing strength of plate:-

$$P_{T1-1} = (B - nd_o) \cdot t \cdot \sigma_{at}$$

$$P_{T2-2} = (B - 3d_o) \cdot t \cdot \sigma_{at} + 3R_V$$

**Strength of Joint:-**

$$\min \left[ \begin{array}{l} nR_v \text{ (strength of all rivets)} \\ P_T \text{ (tensile strength of plate)} \end{array} \right]$$

where n = number of all rivets in each side of joint

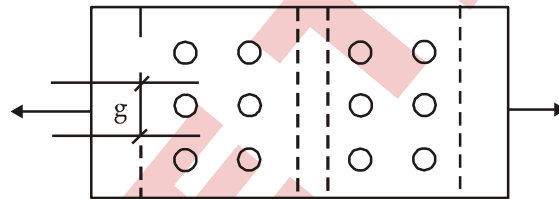
**Efficiency of joint:-**

$$\eta = \frac{\text{Strength of joint } [\min(nR_v, P_T)]}{\text{Tensile strength of solid plate } [Bt\sigma_{at}]}$$

For efficient joint tensile strength or tearing strength of plate should be less than or equal to strength of all rivet.

$$\eta = \frac{(B - nd_o) \times t \times \sigma_{at}}{B \cdot t \cdot \sigma_{at}}$$

$$\eta = \frac{(B - nd_o)}{B}$$

**Note:-**

1. Strength of rivet in per pitch length

$$= nR_v$$

n = Number of rivet in a pitch/gauge length

2. Tearing strength of plate per pitch length

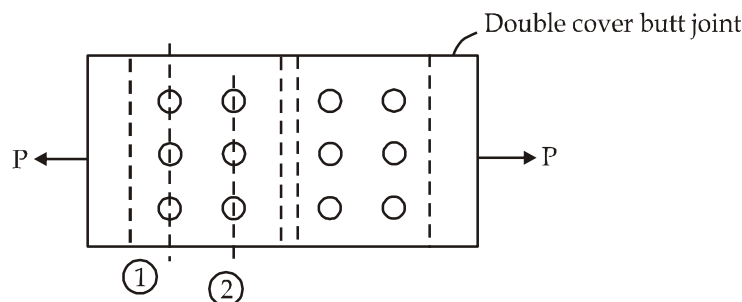
$$= (g - nd_o) \times t \times \sigma_{at}$$

For a efficient joint tearing strength of plate should be less than equal to strength of all rivets.

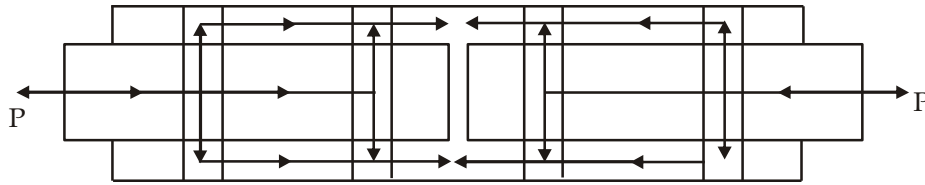
$$(g - nd_o) \times t \times \sigma_{at} \leq nR_v$$

**Arrangement of rivets:-**

1. Chain riveting:-







Load path diagram

Tearing strength of main plate at section 1 - 1

$$P_{T1-1} = (B - 3d_o) \cdot \sigma_{at} \cdot t$$

$$P_{T2-2} = (B - 3d_o) \cdot t \cdot \sigma_{at} + 3R_V$$

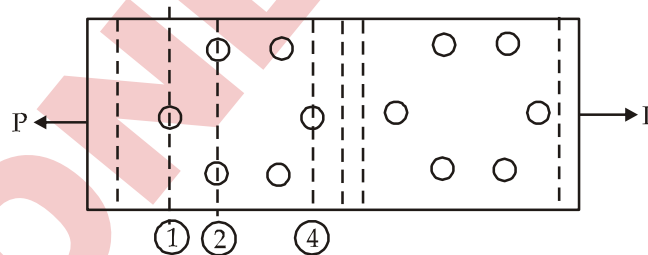
Cover plate strength at section 2 - 2

$$(B_{cover} - 3d_o) \times t \cdot \sigma_{at}$$

t = Summation of thick of cover plates.

- First section will critical for main plate
- Last section will critical for cover plate

## 2. Diamond riveting:-



1. Tearing strength of main plate at section 1 - 1

$$P_{T11} = (B - d_o) \times t \times \sigma_{at}$$

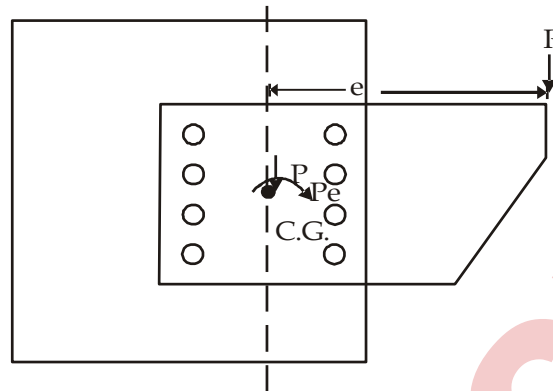
$$P_{T22} = (B - 2d_o) \times t \times \sigma_{at} + R_V$$

Strength of cover plate at section 4 - 4

$$= (B - d_o) \times t \times \sigma_{at}$$

t = Summation of thickness of cover plate.

**Note:-** With the above arrangement both main plate and cover plate carry maximum tensile load.

**Eccentric riveted connection:-****Case I : In plane eccentricity:-**

$e$  = eccentricity

Due to in plane eccentricity a direct force  $P$  and twisting moment of  $Pe$  will be generated at C.G. of rivet group.

**Analysis:-**

1.  $F_1$  = shear due to direct force  
 $P$  at CG

$$F_i = \frac{P}{\sum A_i} \times A_i = F_1 = \frac{P}{\sum A_i} \times A_i$$

If area of all rivets is same

$$F_1 = \frac{P}{nA} \cdot A$$

$$F_1 = \frac{P}{n}$$

$n$  = Number of rivets

Direction  $\downarrow$  (vertically downward)

2. Force due to twisting moment:-

$$T = Pe$$

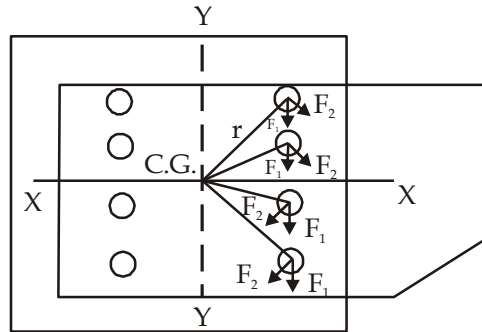
$$\frac{T}{J} = \frac{\tau}{r} = \frac{G\theta}{L}$$

$T$  = Twisting moment

$J$  = Polar moment of inertia

$\tau$  = Shear stress

$r$  = Radial distance of rivet from C.G. of rivet group



$$\frac{T}{J} = \frac{\tau}{r} = \frac{G\theta}{L}$$

$$\tau = \frac{T}{J} \times r$$

$$\text{Force} = \tau \times A$$

$$= \frac{T}{J} \times r \times A$$

$$F_2 = \frac{Pe}{J} \times r \times A \quad [T = P.e]$$

$J$  = Polar moment of inertia

$$J = \sum A_i r_i^2 \Rightarrow A_1 r_1^2 + A_2 r_2^2 + A_3 r_3^2 + \dots + A_n r_n^2$$

$$F_{2i} = \frac{Pe}{\sum A_i r_i^2} \cdot r_i \times A_i$$

$r_i$  = radial distance of rivet from the CG of rivet group.

If area of C/S is same

$$\begin{aligned} & A r_1^2 + A r_2^2 + \dots + A r_n^2 \\ & = A \sum r_i^2 \end{aligned}$$

$$F_{2i} = \frac{Pe}{A \sum r_i^2} \times r_i \times A$$

$$F_{2i} = \frac{Pe}{\sum r_i^2} \times r_i$$

$$\text{Resultant} = \sqrt{F_1^2 + F_2^2 + 2F_1 F_2 \cos \theta}$$

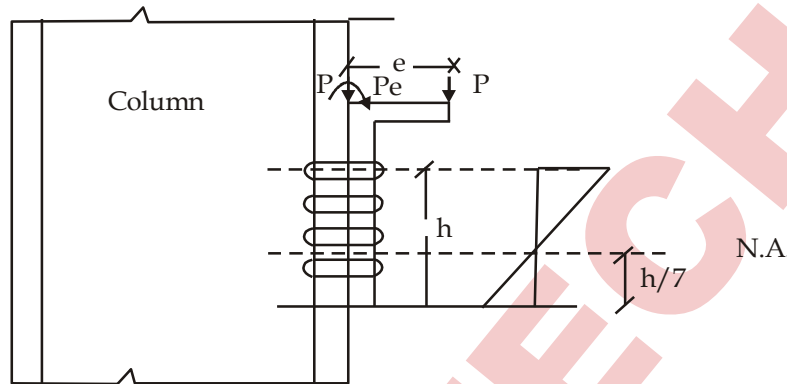
$\theta$  = Angle between  $F_1$  and  $F_2$  measured in anticlockwise from  $F_1$  to  $F_2$

- For Rivet to be safe

$$F_{\text{resultant}} \neq R_v$$

**Note:-** Generally the rivet at maximum distance from the CG of the rivet group and nearest to the load line having maximum force.

**Case II : Out of plane eccentricity:-**



In this arrangement load is out of plane hence B.M will generate in this connection.

Due to this BM rivets above NA will be resisted to tensile force along with direct shearing and bearing.

Rivet below NA will contribute partially in resisting compression because most of the compression effect will be resisted by plate.

Considering both the effect of tension and shear for rivet above NA these interaction equation should be check.

$$\frac{\sigma_{\text{at, calculated}}}{\sigma_{\text{at}}} + \frac{\tau_{\text{va, calculated}}}{\tau_{\text{va}}} \neq 1.4 \text{ (WSM)}$$

$$\left( \frac{P_{\text{at, calculated}}}{P_{\text{at}}} \right)^2 + \left( \frac{P_{\text{s, calculated}}}{P_{\text{s}}} \right) \leq 1 \text{ (LSM)}$$

$\sigma_{\text{at, calculated}}$  = calculated axial tensile stress

$\sigma_{\text{at}}$  = Permissible axial tensile stress

$\tau_{\text{va, calculated}}$  = calculated shear stress

$\tau_{\text{va}}$  = Permissible shear stress

$P_{\text{at, calculated}}$  = calculated axial tensile force

$P_{\text{at}}$  = Safe axial tensile force

$P_{\text{s, calculated}}$  = calculated shear force

$P_{\text{s}}$  = Safe shear force

# CHAPTER - 3

## (BOLTED DESIGN (LSM))

- Unfinished bolt or ordinary bolt or Bearing type bolts.
  - High strength bolt or friction type bolt
- (1) **Unfinished or bearing type bolt:-** These types of bolt are also called common, rough and black bolt. It is used for light structure subjected to static load. It is not suitable for connection subjected to impact or vibrations. Load transfer by shearing and bearing.
- (2) **High strength bolt (friction type):-** These bolt may be tighten until they have very high tensile stresses, two or more time ordinary bolts so that the connected parts are clamped tightly together between the bolts and nut.

In this type of bolts load transfer by friction between connected parts.

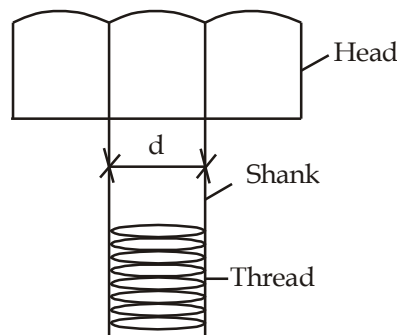
### Design of ordinary bolts:-

#### 1. Grade of bolts:-

Grade	Ultimate Stress	Yield Stress
4.6	$f_{ub} = 400 \text{ N/mm}^2$	$f_{yb} = 400 \times 0.6 = 240 \text{ N/mm}^2$
8.8	$f_{ub} = 800 \text{ N/mm}^2$	$f_{yb} = 800 \times 0.6 = 480 \text{ N/mm}^2$

#### 2. Diameter of bolt hole:-

Diameter of bolt	Diameter of hole
12 - 14 mm	+ 1 mm
16 - 24 mm	+ 2 mm
> 24 mm	+ 3 mm



$$\text{Area of shank} = \frac{\pi}{4} d^2$$

$$\text{Area of threaded portion} = 0.78 \times \frac{\pi}{4} d^2$$

### 3. Shearing strength of bolt:-

For single shear:-

$$V_{dsb} = \text{Resisting area} \times \text{Permissible shear stress}$$

$$V_{dsb} = 0.78 \frac{\pi}{4} d^2 \cdot \frac{f_{ub}}{\sqrt{3} \cdot \gamma_{mb}}$$

$$\gamma_{mb} = 1.25$$

For double shear:-

$$V_{dsb} = 2 \times 0.78 \cdot \frac{\pi}{4} d^2 \cdot \frac{f_{ub}}{\sqrt{3} \times 1.25}$$

### 4. Bearing strength of bolt:- (bearing failure at 6m)

$$V_{dpb} = 2.5 K_b \times d \times t \times \frac{f'_{ub}}{1.25}$$

$f'_{ub}$  = minimum ultimate tensile stress of plate and bolt material

$d$  = diameter of bolt

$t$  = thickness of thinner member connected

$$K_b \text{ :- minimum of } \begin{cases} \frac{e}{3d_0} \\ \frac{p}{3d_0} - 0.25 \\ \frac{f_{ub}}{f_u} \\ 1 \end{cases}$$

$e$  = end distance

$d_0$  = diameter of hole

$p$  = Pitch

$f_{ub}$  = Ultimate stress of bolt

$f_u$  = Ultimate stress of plate

Bolt value ( $V_{db}$ ) = strength of a bolt

$$\text{Min} \begin{cases} V_{dsb} \\ V_{dpb} \end{cases}$$

**Strength of plate:-**

#### 1. Solid plate strength (Gross area strength):-

$$T_{dg} = A_g \cdot \frac{f_y}{1.10} \text{ [serviceability criteria]}$$



## 2. Net area strength (Limit state of collapse):-

$$T_{dn} = A_{net} \cdot \frac{f_u}{1.25} \times 0.9$$

Tensile strength of plate is ( $T_{dp}$ )

$$\text{Min} \begin{cases} T_{dg} \\ T_{dn} \end{cases}$$

Strength of Joint:-

$$\text{Min} [ T_{dn}, T_{dg}, V_{dsb}, V_{dps} ]$$

Efficiency of Joint:- Strength of Joint/Strength of solid plate

$$\frac{\text{Min} [ T_{dn}, T_{dg}, V_{dsb}, V_{dps} ]}{A_g \times \frac{f_y}{1.1}}$$

## 3. Gross area strength of plate:-

$$T_d = A_g \times \frac{f_y}{1.10}$$

### Friction bolt or High strength or HSFG - High strength friction grip bolts

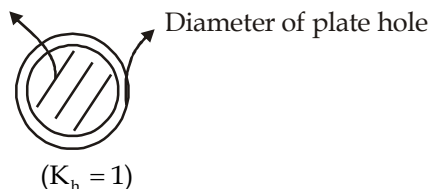
- These bolts provide extremely efficient connection in dynamic load condition.
  - These bolt should be tighten to their proof load.
  - The HSFG connections are designed such that under service load the force does not exceed the frictional resistance.
  - When the external force exceed the frictional resistance, the bolt come into contact with the plate and start bearing.
- (1) Slip critical connection (slip between plates is not allowed)
- Bolts are designed on the basis of slip resistance.

$$\text{Design slip resistance of bolt } (v_{dsf}) = \frac{\mu_f \times K_h \times \eta_e \times F_o}{\text{FOS}}$$

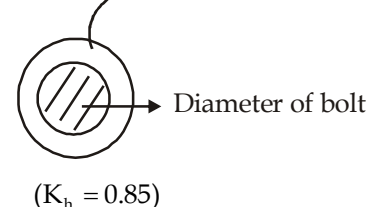
$\mu_f$  = Slip factor depends upon surface treatment (if not given then assume =0.5)

$K_h$  = depends upon type of bolt hole

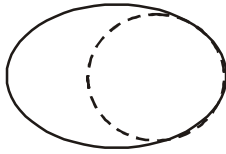
Diameter of bolt



Diameter of bolt hole



(a) Clearance hole or standard hole

 $(K_h = 0.85)$ 

(b) Over sized hole

 $(K_h = 0.7)$ 

(c) Short slotted hole

(d) Long slotted hole

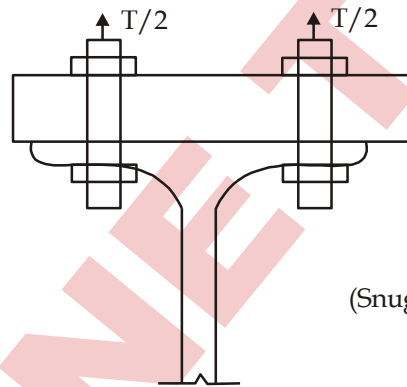
 $f_o$  = tensile strength at proof stress

$$= 0.78 \cdot \frac{\pi}{4} \cdot d^2 f_o$$

 $f_o$  = proof stress =  $0.7 \times f_{ub}$   $f_{ub}$  = ultimate tensile stress of bolt**FOS:-** If slip resistance is designed for service load FOS = 1.10

If slip resistance is designed for ultimate load FOS = 1.25

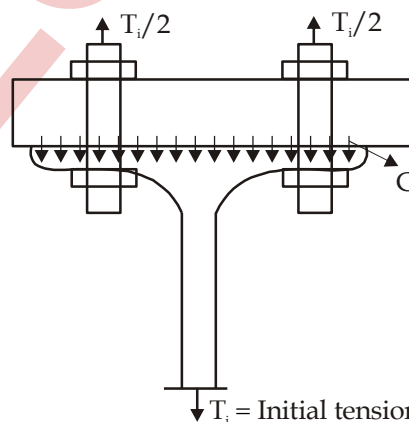
(2) Slip permitted connection:- Design same as bearing bolt.

**Prying Force:-**

(Snug type bolt)

 $T$  = Externally applied load

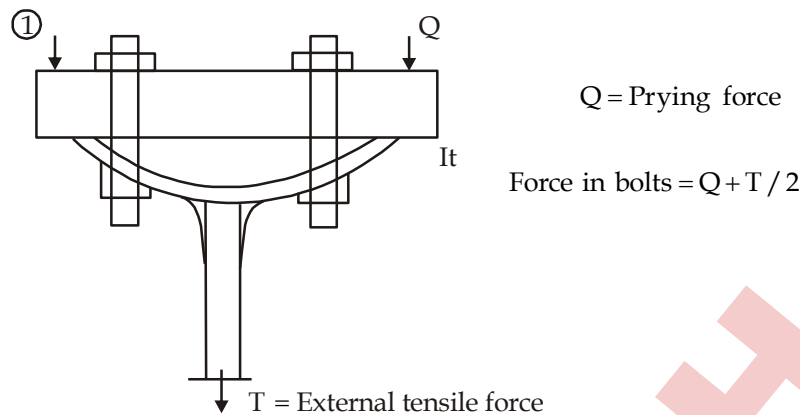
(i) Bearing type bolt



Clamping between plates

 $T_i$  = Initial tension due to pretensioned

(ii) Friction type bolt



1. **Bearing type connection:-** In this connection before any external tension applied, the force in the bolt is almost zero since the bolts are snug type.

As the external tension is increased it is equilibrated by the increase in bolt tension depending on the relative flexibility of the plate and the bolt. Sometimes the opening of the joint may be accomplished by prying action. This prying action increases tensile force in bolts.

**HSFG bolt:-** It is seen that even before any external load is applied, the force in the bolt is equal to proof load due to this there is clamping force between the plates in contact.

- When the external load is applied part of the load (approx 10%) is equilibrated by the increase in bolt force. The balance of the force is equilibrated by the reduction in contact between the plates.

This process is continuous until the contact between the plates is maintained.

- After the external force exceeds this level the behaviour of the bolt under tension is essentially the same as that in a bearing type bolt.
- It is assumed that the load on the joint is distributed equally among on the bolt.
- If length of joint increases 15 times of diameter of bolt (15d) then the force shared by the end bolts may be so high that it may leads to progressive joint failure which is called unbuttoning effect.
- When length of joint exceed 15d then shear capacity of bolt is reduced by reduction factor.

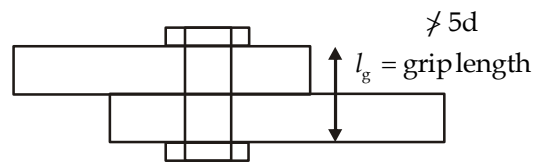
$$\beta_{ij} = 1.075 - \frac{l_j}{200d}$$

$l_j$  = length of joint

d = diameter of bolt.

$$0.75 \leq \beta_{ij} \leq 1$$

## 2. Grip Length:-



- If grip length is increased then the efficiency of joint decreased because of additional bending stresses develop in the bolt.

If grip length exceed  $5d$  then a reduction factor is taken while calculating shear strength of bolt.

$$\beta_{lg} = \frac{8d}{3d + l_g} \quad [l_g > 5d]$$

$l_g$  = Grip length

[In any case  $l_g \neq 8d$ ]

# CHAPTER - 4

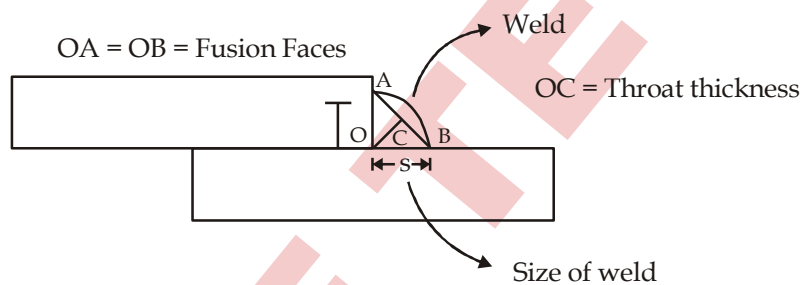
## (WELDED CONNECTION)

Welding is a process of making joint between two similar metals by fusing them together at very high temperature ( $3600^{\circ}\text{C}$ ) or greater than melting point to produce similar section after solidification.

**Why welding? or Welding as a connection process:-**

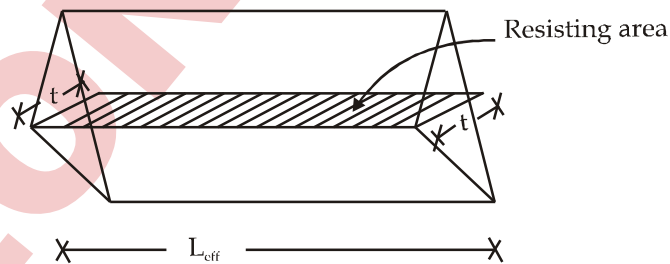
1. No deductions are there for holes thus the gross section is effective in carrying loads.
2. Welding is the only process that produces a one piece connection.
3. Welded joints are better for impact loads and vibrations.

### 1. Fillet Weld:-



Throat thickness,  $t = \frac{S}{\sqrt{2}} = 0.707 S$

Resisting area = Throat thickness  $\times$  Effective length of weld

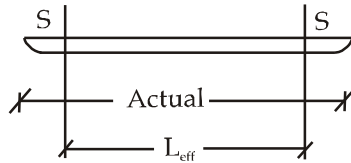


### 2.

Angle between Fusion faces	Throat thickness (S= size of weld)
$60^{\circ} - 90^{\circ}$	0.7 S
$91^{\circ} - 100^{\circ}$	0.65 S
$101^{\circ} - 106^{\circ}$	0.6 S
$107^{\circ} - 113^{\circ}$	0.55 S
$114^{\circ} - 120^{\circ}$	0.5 S

### 3. Effective length of weld:-

$L_{\text{eff}}$  = actual length of weld -  $2 \times S$  where  $S$  = size of weld



- Actual length of weld should not be less than 4 times of size of weld.
4. Minimum size of weld:- Depends upon the thickness of thicker plate connected.

Plate thickness	Minimum size of weld
Upto 10 mm	3 mm
11 to 20 mm	5 mm
21 to 32 mm	6 mm
> 32 mm	8 mm

- Depends
5. Maximum size of weld:- Depends upon thickness of thinner plate.

$$t_2 > t_1$$

- Maximum size of weld = thickness of thinner plate - 1.5 mm (for square edge)
  - $3/4$  thickness of thinner plate (rounded edge)
6. Strength of weld:-

LSM:-

$$F_{\text{wd}} = \text{Resisting area} \times \text{Permissible stress}$$

Resisting Area = throat thickness ( $t$ )  $\times$  Effective length of weld  $\times$

$$f_{\text{wd}} = \frac{f_u}{\sqrt{3} \times 1.25}$$

$$= t \times l_{\text{eff}} \times \frac{f_u}{\sqrt{3} \times 1.25}$$

FOS:- 1.5 field welded

= 1.25 Shop weld

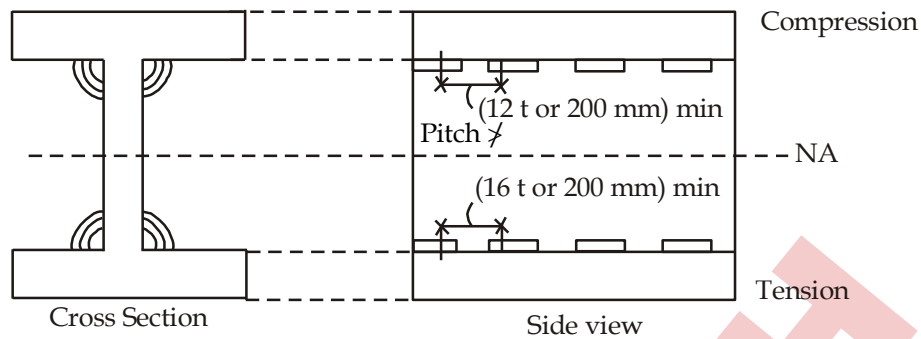
WSM:-

$$P_{\text{wd}} = l_{\text{eff}} \times t \times f_s$$

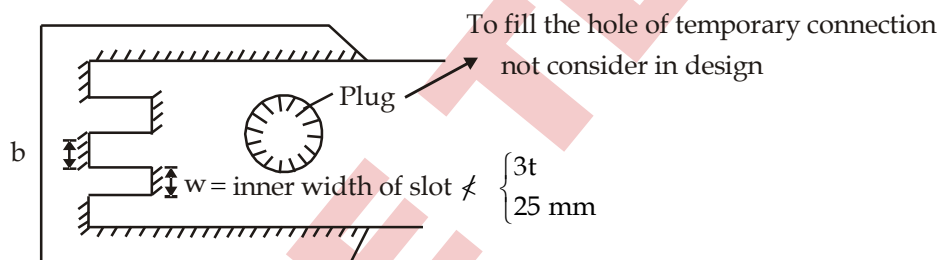
$$f_s = 0.4 F_y \text{ or } 110 \text{ N/mm}^2$$

$$f_y = 250 \text{ N/mm}^2$$



**Intermittent fillet weld:-**

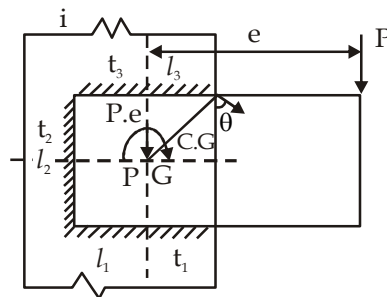
Intermittent fillet weld are provided to transfer calculated stress across the joint when the strength required is less than the developed by a continuous fillet weld.

**Plug and Slot welding:-**

$t = \text{thickness of plate connected}$

$b = \text{outer width of slot} \neq 2t$

→ Slot weld is used when length of overlap is restricted or limited.

**Eccentric loaded connection****Type-1:- In plane eccentricity**

Due to eccentric load  $P$ , C.G. of rivet group subjected a direct load  $P$  and twisting moment  $Pe$

$F_1$  = stress in weld due to direct force P

$$F_1 = \frac{P}{\text{Prestressing area}}$$

$$F_1 = \frac{P}{l_{\text{eff}1} \times t_1 + l_{\text{eff}2} \times t_2 + l_{\text{eff}3} \times t_3}$$

$F_2$  = shear stress due to twisting moment Pe

$$\frac{T}{J} = T/r = F_2 = \frac{T}{J} \times r$$

J = Polar moment of inertia

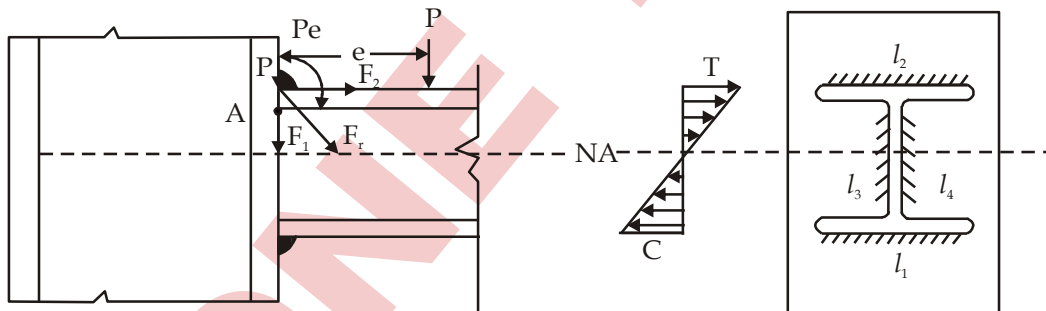
r = radial distance of point from C.G. of weld group.

$$F_r = \sqrt{F_1^2 + F_2^2 + 2F_1F_2 \cos \theta}$$

$$F_r \leq \text{or } \neq 108 \text{ N/mm}^2 \text{ (WSM)}$$

$$F_r \leq \text{or } \neq \frac{f_u}{\sqrt{3} \times \gamma_{mw}} \text{ (LSM)}$$

Out of plane eccentricity:-



$$F_1 = \frac{P}{\text{Prestressing area}}$$

$$F_1 = \frac{P}{l_1 t_1 + l_2 t_2 + l_3 t_3 + l_4 t_4}$$

$F_2$  = Horizontal shear stress at point A due to bending tension

$$F_r = \sqrt{F_1^2 + F_2^2} \leq 108 \text{ N/mm}^2 \text{ {WSM}}$$

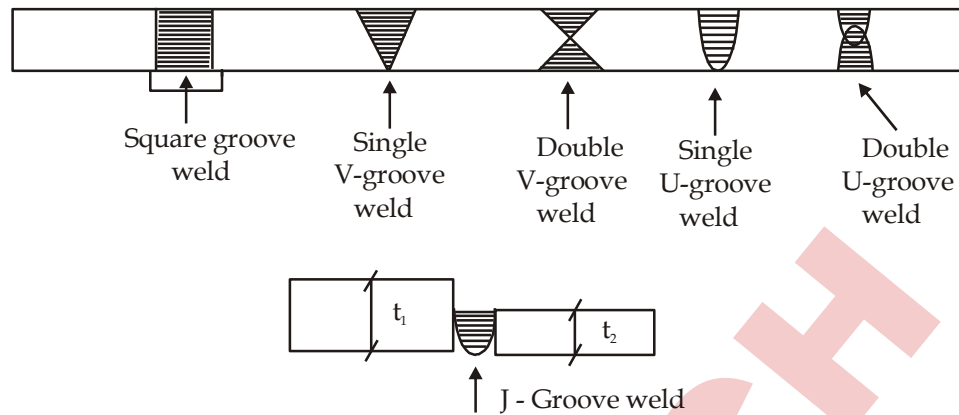
$$\frac{M}{I_{xx}} = F/y_{\text{max}}$$

$$F_2 = \frac{M}{I_{xx}} y_{\text{max}}$$

$$F_r = \sqrt{3F_1^2 + F_2^2} \leq \frac{f_u}{\sqrt{3} \times \text{FOS}} \text{ {LSM}}$$

### Designing of Butt weld or Groove weld

→ Used when the member to be jointed are lined up.



#### Partially Penetration of weld material

Single V-Groove  
Single U-Groove  
J-weld

#### Fully Penetration of weld material

Double V-Groove  
Double U-Groove  
Square-weld

Groove welds are usually design for direct tension or compression

#### Design strength of groove weld:-

It designs given by tension or compression

$$P_{Tw} = \frac{l_{eff} \times t_e \times f_{yw1}}{FOS}$$

$l_{eff}$  = effective length of weld

$$t_e = \text{For fully penetration of weld} = \min \left\{ t_1, t_2 \right\}$$

$$t_e = \text{For partially penetration of weld} = \min \left\{ \frac{5}{8} t_1, \frac{5}{8} t_2 \right\}$$

$$f_{yw1} = \min \left\{ \begin{array}{l} f_y = \text{yield stress of plate material} \\ f_{yw} = \text{yield stress of weld material} \end{array} \right.$$

$$FOS = 1.5 \text{ [Field weld]} \\ = 1.25 \text{ [Shop weld]}$$

If design govern by shear

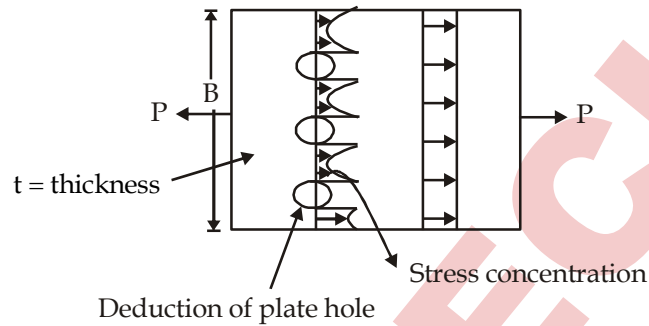
$$f_{yw2} = \min \left\{ \begin{array}{l} \frac{f_y}{\sqrt{3}} \\ \frac{f_{yw}}{\sqrt{3}} \end{array} \right.$$

$$P_{Tw} = \frac{l_{eff} \times t_e \times f_{yw2}}{FOS}$$

# CHAPTER - 5

## (DESIGN OF TENSION MEMBER)

### Design of Pitch (P)



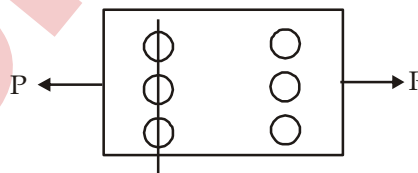
$$\begin{aligned} \text{Net area} &= A_{\text{gross}} - \text{deduction of hole} \\ &= B \times t - n d_o t \\ &= (B - n d_o) \times t \end{aligned}$$

The Net section area of a tension member is the gross section area of the member section minus area of the maximum number of holes.

The unit stress in a tension member is increased due to the presence of a hole this is because the area of steel to which load is distributed is reduced and some concentration of stress occurs along the edges of hole.

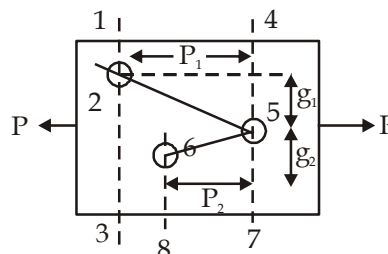
Net section area of plate elements

A. If rivet/Bolt holes in same line



$$\begin{aligned} A_{\text{net}} &= Bt - 3d_o \times t \\ &= (B - 3d_o) \times t \end{aligned}$$

B. .... are staggered



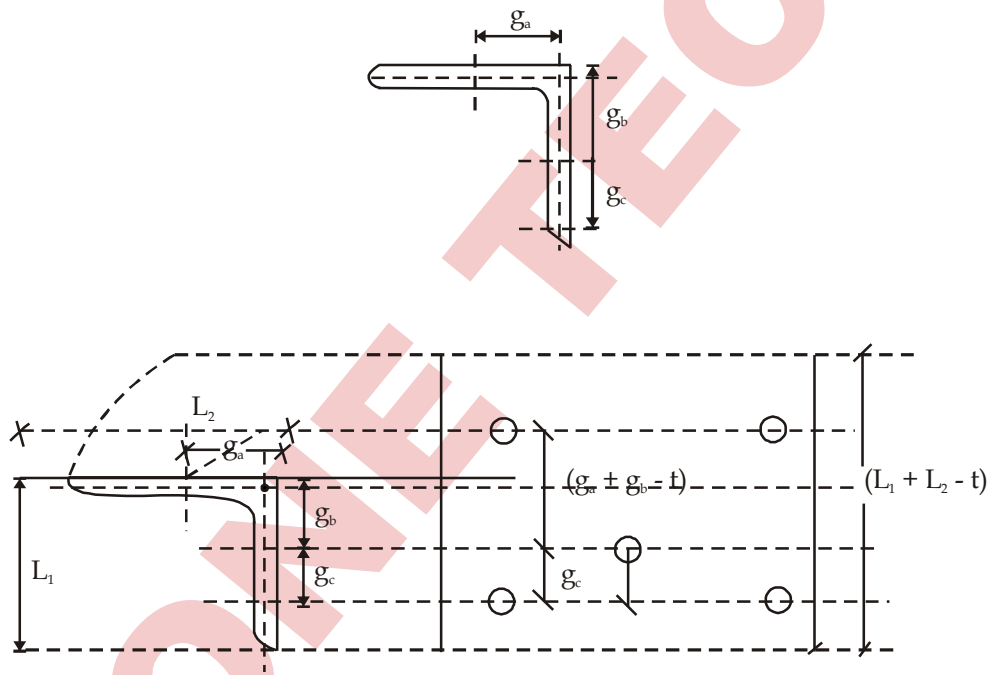
$$\text{Deduction} = n d_o \times t - \frac{\sum P_i^2}{4g_i} \times t$$

$$A_{\text{net}} = Bt - n d_o t + \frac{\sum P_i^2}{4g_i} t$$

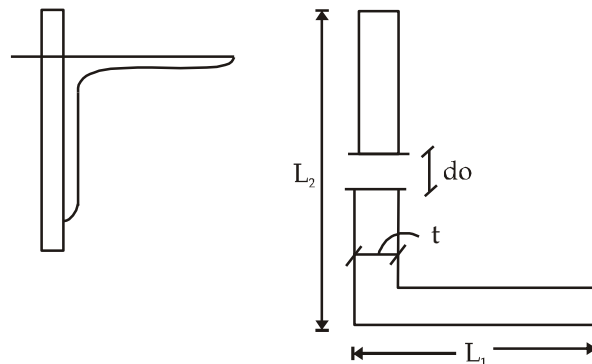
$$= \left[ (B - n d_o) + \frac{\sum P_i^2}{4g_i} \right] \times t$$

## 2. Net sectional area for angle section:-

### A. When all legs are connected:-



### B. If only one leg is connected:-



$$\text{Net area of connected leg} = (L_2 - t/2 - d_o) \times t$$

$$\text{Net area of outstanding leg} = (L_1 - t/2) \times t$$

$$A_{\text{net}} = \text{net area of angle section}$$

$$= (L_2 - t/2 - d_o) t + (L_1 - t/2)t$$

Effective net area of angle section:- (LSM)

$$A_{\text{net effective}} = K_1 K_2 K_3 K_4 A_{\text{net}}$$

$K_1$  = ductility factor as per IS800 - 2007

$K_2$  = depends upon method of fabrication

$K_2 = 1$  (For drilled hole)

$K_2 = 0.85$  (For punched hole)

$K_3$  = Geometory factor = 1

$K_4$  = Shear leg factor

$K_4 = 0.6$  = If number of bolt = 2

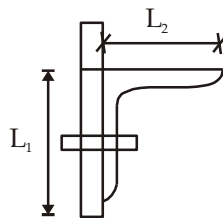
0.7 = If number of bolts = 3

0.8 = If number of bolts  $\geq 4$

0.8 = If leg is connected by welding

**WSM approach:-**

1. If single angle connected to gusset plate:-



$$A_{\text{net}} = A_1 + A_2$$

$$A_1 = (L_1 - d_o - t/2) \times t$$

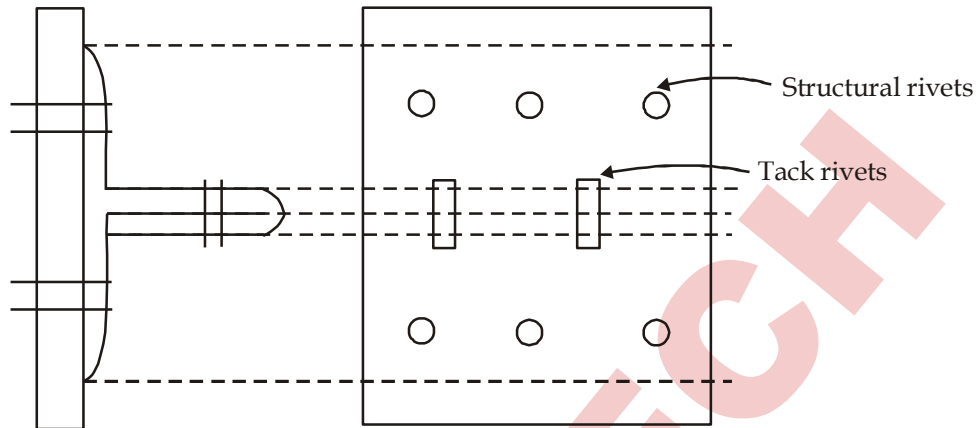
$$A_2 = (L_2 - t/2) \times t$$

$$A_{\text{net}} = A_1 + K A_2$$

$$K = \frac{3A_1}{3A_1 + A_2}$$



2. If two angles are connected to gusset plate
- (i) Both angles are on same side of gusset plate



- (a) If both angles are tack riveted through

$$A_1 = (L_1 - d_o - t/2) \times t] \times 2$$

$$A_2 = (L_2 - t/2) \times t] \times 2$$

$$A_{\text{net}} = A_1 + KA_2$$

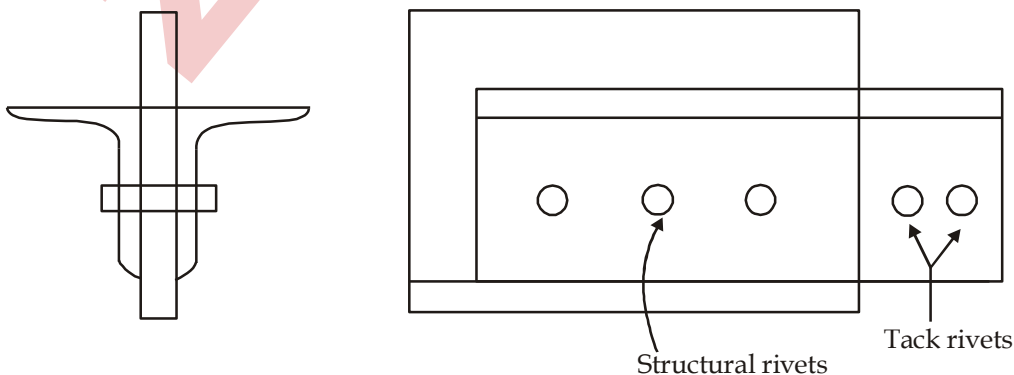
$$K = \frac{5A_1}{5A_1 + A_2}$$

- (b) If both angles are not tack riveted

$$A_{\text{net}} = A_1 + KA_2$$

$$K = \frac{3A_1}{3A_1 + A_2}$$

- (ii) Both angles are on either side of gusset plate:-



- (a) If both angles are tack riveted:-

$$A_{\text{net}} = A_1 + KA_2$$

$$K = 1$$

- (b) If both angles are not tack riveted:-

$$A_{\text{net}} = A_1 + KA_2$$

$$K = \frac{3A_1}{3A_1 + A_2} \text{ (same as single angle)}$$

### Design strength of tension member:-

- (i) Plate or Flats:-

- (a) Gross Area strength:-

$$P_{Tg} = \frac{A_{\text{gross}} \times f_y}{1.1}$$

- (b) Net section Area strength:-

$$P_{Tn} = \frac{0.9A_{\text{net}} \times f_u}{1.25}$$

$$\text{Strength} = \text{Min} [P_{Tg}, P_{Tn}]$$

- (ii) If one leg of angle section connected:-

- (a) Gross Area strength:-

$$P_{Tg} = A_{\text{gross}} \times \frac{f_y}{1.10}$$

- (b) Net area strength:-

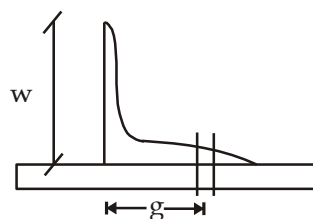
$P_{Tn}$  = Tearing strength of angle section

$$= 0.9 A_{\text{net connected}} \times \frac{f_u}{1.25} + \beta \times A_{\text{gross outstanding}} \times \frac{f_y}{1.10}$$

$$\beta = 1.4 - 0.076 \times \frac{w}{t} \times \frac{f_y}{f_u} \times \frac{b_s}{L_c} \geq 0.7 \leq \frac{f_u}{f_y} \times \frac{1.10}{1.25}$$

$A_{nc}$  = Net area of connected leg

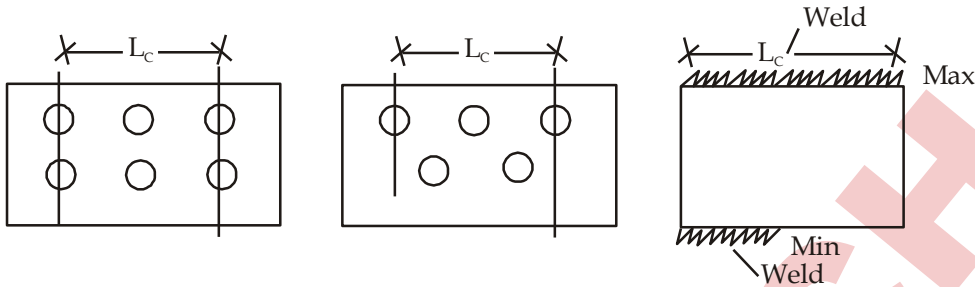
$A_{go}$  = Gross area of outstanding leg



$w$  = length of outstanding leg

$b_s = w + g - t$

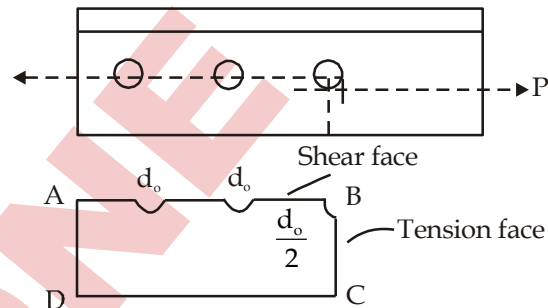
$L_c$  = distance between end bolts



$L_c$  = distance between end bolts or maximum length of weld =  $L_c$

**Alternate Approach:-**

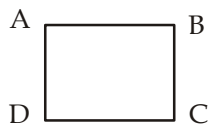
$$P_{Tn} = \frac{K_1 K_2 K_3 K_4 \times A_{net} \times 0.9 \times f_u}{1.25}$$



## 2. Tension yielding and shear fracture

$$P_{Tb1} = \frac{A_{Tg} \times f_y}{1.10} + \frac{A_{vn} \times 0.9 \times f_u}{\sqrt{3} \times 1.25}$$

$A_{Tg}$  = Gross area of tensile force =  $BC \times t$



$A_{vn}$  = Net area of shear force

$$= (AB - d_o - d_o - d_o/2) \times t$$

(ii) Shear yielding and tension fracture:-

$$P_{Tb_2} = \frac{A_{vg} \times f_y}{\sqrt{3} \times 1.10} + \frac{0.9 \times A_{Tn} \times f_u}{1.25}$$

$A_{vg}$  = Gross area of shear force

$$= AB \times t$$

$A_{Tn}$  = Net area of tensile force

$$= (BC - \frac{d_o}{2}) \cdot t$$

Block shear strength

$$= \text{Min} \left\{ \begin{array}{l} P_{Tb_1} \\ P_{Tb_2} \end{array} \right.$$

**Tensile strength of member:-**

$$= \text{Minimum of } [ P_{Tg}, P_{Tn}, P_{Tb_1}, P_{Tb_2} ] \text{ (LSM)}$$

**Tensile strength (WSM):-**

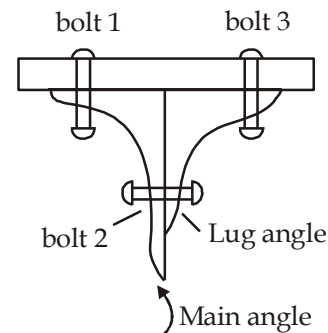
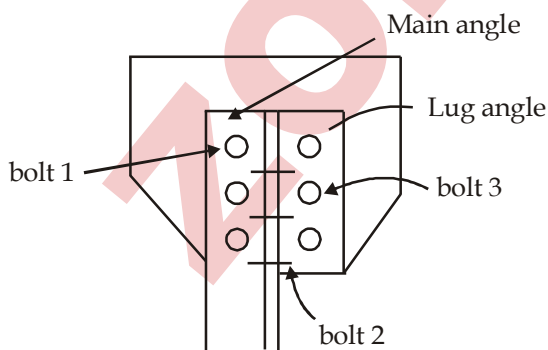
$$P_T = A_{net} \times 0.6 F_y$$

or

$$A_{net} \times \sigma_{at}$$

$\sigma_{at}$  = Permissible stress in axial tension

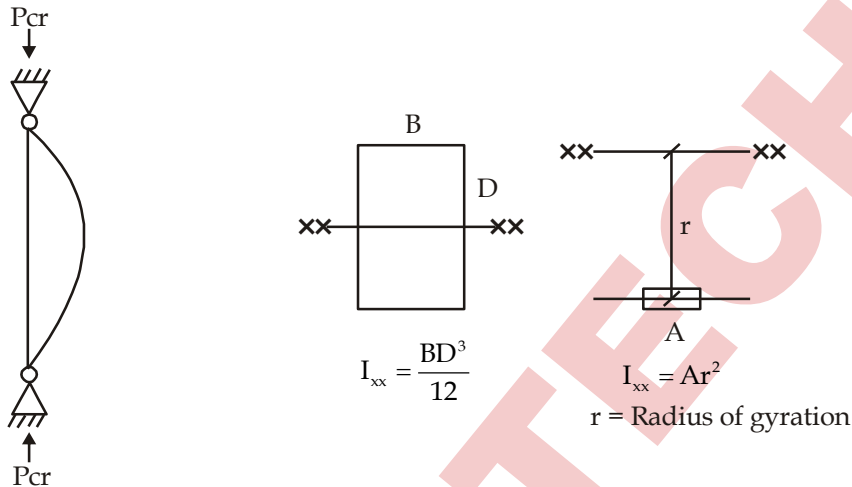
**Lug angle:-**



- Lug angle is a small piece of angle used to connect outstand leg of the member to gusset plate.
- If lug angle is used the outstand leg of the main angle behaves like a connected leg and it reduces the shear leg effect.

# CHAPTER - 6

## (DESIGN OF COMPRESSION MEMBER)



### Euler Buckling Formula:-

$$P_{cr} = \frac{\pi^2 EI}{L_{eff}^2}$$

$$I = Ar^2$$

where  $r$  = radius of gyration

$A$  = cross-x area

$$r = I/A$$

$$P_{cr} = \frac{\pi^2 EAr^2}{L_{eff}^2}$$

$$P_{cr} = \frac{\pi^2 EA}{(L_{eff}/r)^2} \quad \lambda = L_{eff}/r \quad \text{where } r = \text{appropriate radius of gyration}$$

$$\frac{P_{cr}}{A} = \frac{\pi^2 E}{(L_{eff}/r)^2} \quad \lambda = \text{Slenderness ratio}, \quad l_{eff} = \text{effective length}$$

$$f_{cr} = \frac{\pi^2 E}{\lambda^2}$$

$f_{cr}$  = Critical buckling stress

For  $F_{cr}$  minimum

$\lambda$  should be maximum

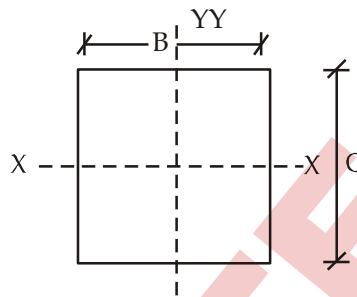
$\lambda \uparrow$  then  $r$  should be minimum

Section should be buckle about minimum radius of gyration

(For  $r$  min)

**Case I:-**

If one of the centroidal axis is axis of symmetry.



$$I_{XX} = BD^3/12$$

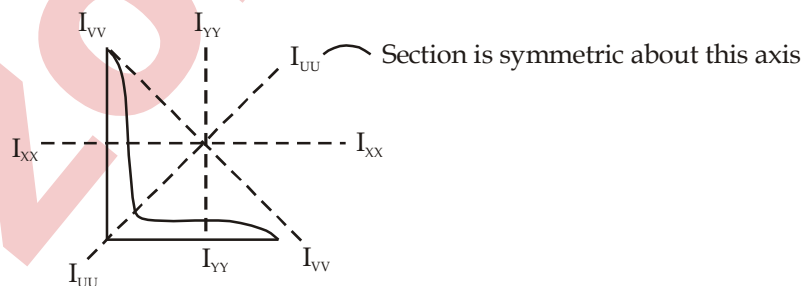
$$I_{YY} = DB^3/12$$

$I_{YY}$  is minimum so buckling is about  $I_{YY}$ .

The section symmetrical about both the centroidal axis or even with one axis of symmetry will buckle about one of the centroidal axis.

**Case II:-**

If none of the centroidal axis is not axis of symmetry.

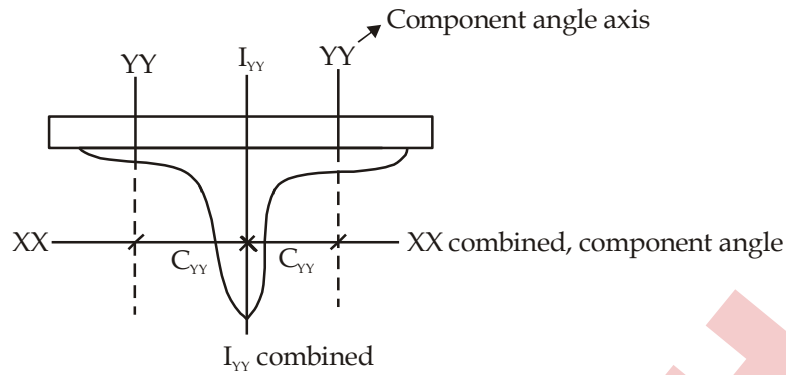


$I_{UU}$  and  $I_{VV}$  are principle axis.

The section is asymmetrical about the centroidal axis will buckle about the principle axis for which radius of gyration will be minimum.

**Case III:-**

If two angles are placed same side of gusset plate and tack riveted throughout the length.



$I_{XX}$  combined:-

$$= I_{XX} \text{ component} \times 2$$

$I_{YY}$  combined:-

$$= 2 [I_{YY} \text{ component} + A \cdot C_{YY}^2]$$

Minimum [ $I_{XX}$  combined or  $I_{YY}$  combined]

Section will be buckle about minimum of  $I_{XX}$  combined or  $I_{YY}$  combined.

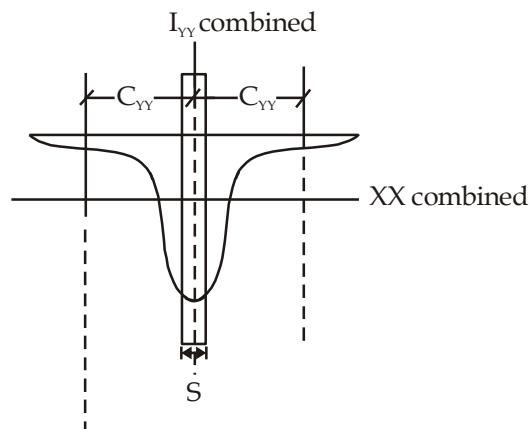
$r_{XX}$  is minimum moment of inertia-

$$\begin{aligned} r_{XX} \text{ component} &= \sqrt{\frac{I_{XX} \text{ combined}}{A \text{ combined}}} \\ &= \sqrt{\frac{2 \times I_{XX} \text{ component}}{2 \times \text{Area component}}} \\ &= \sqrt{\frac{I_{XX} \text{ component}}{\text{Area component}}} \end{aligned}$$

$$r_{XX} \text{ combined} = r_{XX} \text{ component}$$

**Case IV:-**

If two angle section are placed either side of gusset plate.





$I_{XX} \text{ combined} = 2 \times I_{XX} \text{ component}$

$I_{YY} \text{ combined} = 2 \times [I_{YY} \text{ component} + A \times (C_{YY} + S/2)^2]$

→ Minimum slenderness ratio for structural member:-

1. Tension member subjected to live load  $\lambda = 400$
2. Tension member in which reversal of load due to load other than wind or earthquake forces  $\lambda = 180$ .
3. Compression member subjected to live load  $\lambda = 180$ .
4. Tension member reversal of stresses due to wind or earthquake forces  $\lambda = 350$ .
5. Member subjected to compressive forces only due to earthquake and wind  $\lambda = 250$ .
6. Compression flange of beam restrained against lateral torsional buckling  $\lambda = 300$

### Design column formulas:-

**IS 800-1984 - (WSM):-** This code recommended the use of MERCHANT RENKINE'S formula.

$$\sigma_{ac} = \frac{F_{cc} \times F_y \times 0.6}{[(F_{cc})^n + (F_y)^n]^{1/n}}$$

$f_y$  = yield stress

$f_{cc}$  = critical buckling strength

$\eta$  = imperfection factor

$\eta = 1.4$

**IS 800-2007:-** Recommended the use of suitable column curve based upon PERRY ROBERISON approach.

1. Design strength of compression member by WSM approach:-

Step I:- Find out  $\lambda = \frac{L_{eff}}{r_{appropriate}}$

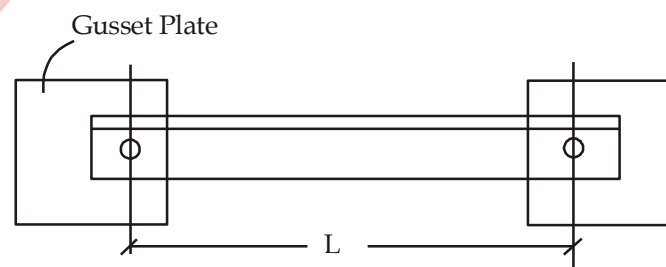
Step II:- By the use of  $\lambda$  value of  $\sigma_{ac}$  will be find out by use of IS-800-1984.

Step III:- Design strength of compression member

$$P_{cd} = \sigma_{ac} \times A_{gross}$$

### Effective Length of Strut:-

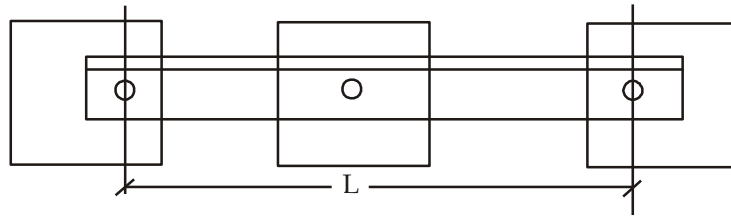
1. **Discontinuous strut:-**



$L_{eff} = L$

Single angle connected by one bolt or rivet.

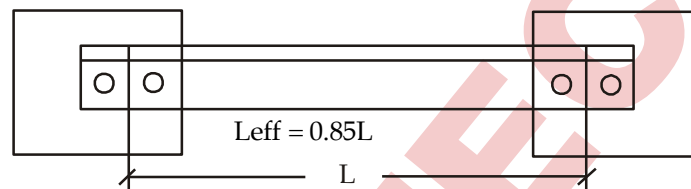
## 2. Continuous strut:-



$$L_{eff} = 0.7L \text{ to } L$$

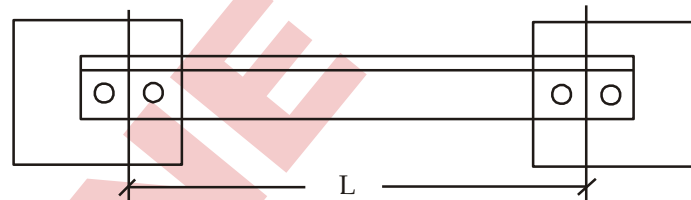
Single or double angle connected with one or more bolt and weld.

## 3. Discontinuous strut:-



Single angle with two or more bolt or weld.

## 4. Discontinuous strut:-



$$L_{eff} = 0.7L \text{ to } 0.85L$$

Double angle with one or more than one bolt or weld.

**Specifications for tack rivets:-** If tack rivets are used to connect components of the column the slenderness ratio for each component between tack rivet.

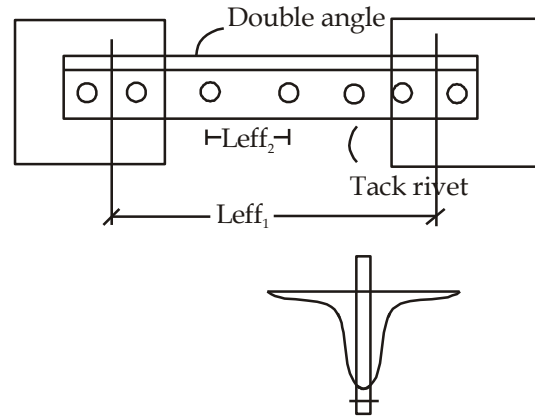
$\lambda$  component between two tack rivets

$$\neq 40$$

$$\neq 0.6 \lambda_{\text{whole}}$$

$\lambda_{\text{whole}}$  OR  $\lambda_{\text{combined}}$  :-

$$= \frac{L_{eff_1}}{\gamma_{xx} \text{ component}}$$



$\lambda$  component between two tack rivets:-

$$= Leff_2 / \text{Min} (r_{VV} \text{ or } r_{UU})$$

**Strength of column according to LSM:-**

- Design  $f_{cd}$  can be calculated as per clause no - 7.1.2.1 of IS 800 - 2007.  
It can also be calculated by table no. 8/ table no. 9 of IS 800-2007.
- The value of  $F_{cd}$  depends upon slenderness ratio and buckling curve of given section.  
Alternate approach to calculate  $f_{cd}$  as per IS 800 - 2007 clause no. 7.1.2.1

$f_{cd}$  = design compressive strength

$$F_{cd} = \frac{F_y}{1.10} \times \eta \quad \eta = \text{Reduction factor}$$

$$\eta = \frac{1}{\phi + [\phi^2 - \lambda^2]^{0.5}}$$

$$F_{cd} = \frac{F_y / 1.10}{\phi + [\phi^2 - \lambda^2]^{0.5}}$$

$$\phi = 0.5[1 + \alpha(\lambda - 0.2) + \lambda^2]$$

$\lambda$  = Non dimensional slenderness ratio

=  $\sqrt{f_y / f_{cc}}$  - critical buckling strength

$\alpha$  = Imperfection factor depends upon buckling curves.

Buckling curve	a	b	c	d
$\alpha$	0.21	0.34	0.49	0.76

**Design of compression member:-** When load is given and size of column is required.

Step I:- Assumptions of  $\sigma_{ac}$  or  $f_{cd}$

- (a) For hot rolled sections = 60 MPa - 80 MPa
- (b) For built up section = 110 MPa - 125 MPa

Step II:- Area required for column section:-

$P =$  Load which is to be resisted

$\sigma_{ac}$  assumed.

Step III:- According to this area required choose a appropriate area from steel table SP-6

Step IV:- After selecting area

$r_{xx}$  and  $r_{yy}$  will be find out

and for  $r_{min}$

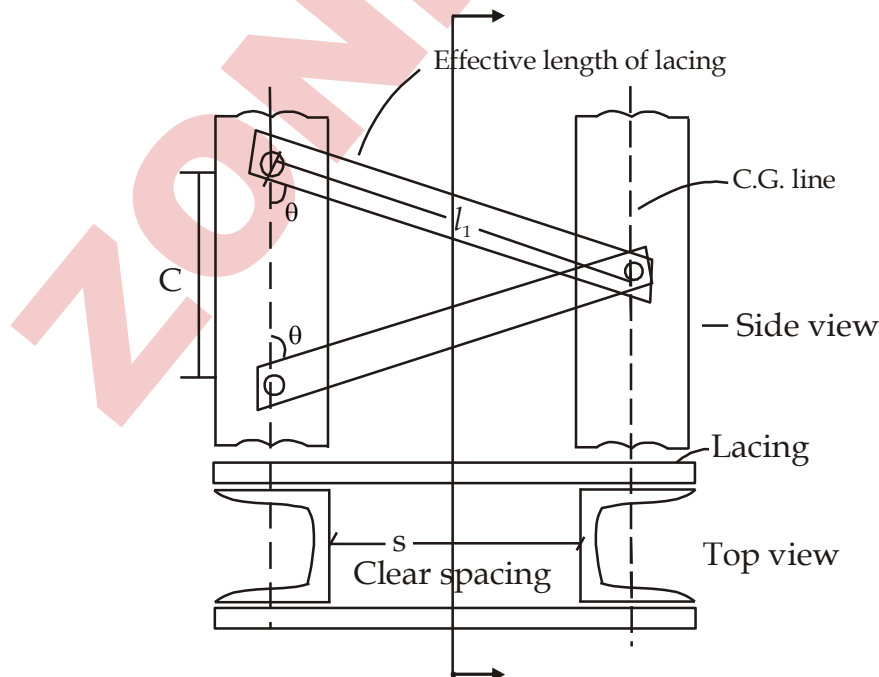
$\lambda = Leff / r_{min}$  will be find out

Step V:- According to this  $\lambda$

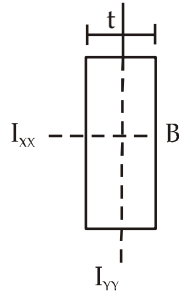
$\sigma_{ac}$  and  $F_{cd}$  will be find out and load carrying capacity of section is determined.

$P_{cd} = \sigma_{ac} \times A_{gross} > P =$  Load for which compression member is to be designed.

**Design of Lacing system:-**



1. Maximum slenderness ratio of lacing flat - 145
2. Angle from vertical should not less than  $40^\circ$  and should not be greater than  $70^\circ$
3.  $\lambda_{\text{Lacing}} = L_{\text{eff}} / r_{\text{min}}$



$$I_{YY} = I_{\text{min}}$$

$$= \frac{Bt^3}{12}$$

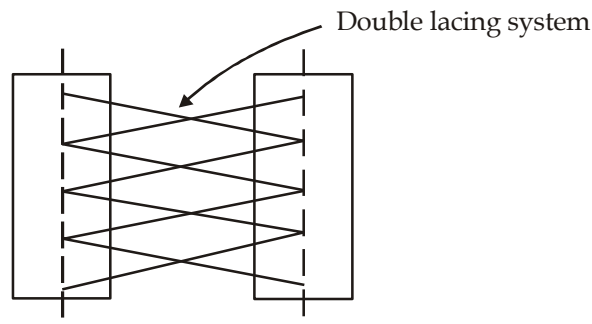
$$r_{\text{min}} = \sqrt{\frac{Bt^3/12}{Bt}}$$

$$\lambda_{\text{Lacing}} = t / \sqrt{12}$$

4. Minimum width of lacing bar

Nominal diameter of rivet	Width
16	50
18	55
20	60
22	65 mm

**Double lacing system:-**



The thickness of lacing bar:-

$$\frac{l_1}{40} = \text{For single lacing}$$

$$\frac{l_1}{60} = \text{For double lacing}$$

To prevent local buckling of column component between any two lacing.

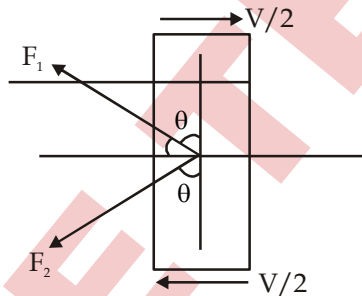
$$\lambda_{\text{component}} = \frac{C}{r_{\text{component}}} \not\geq 50$$

$$\not\geq 0.7\lambda_{\text{whole}}$$

### Forces in lacing bar:-

Lacing system is designed for transverse shear of magnitude 2.5% of column axial load.

To take care of accidental load and initial eccentricity.



$$\sum f_x = 0$$

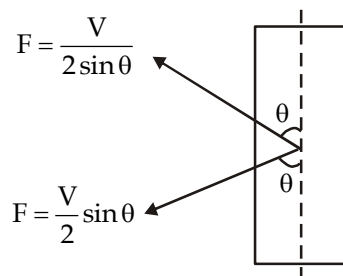
$$F_1 \sin \theta = V/2$$

$$F_1 = V/2 \sin \theta \quad (\text{Tension})$$

$$F_2 \sin \theta + V/2 = 0$$

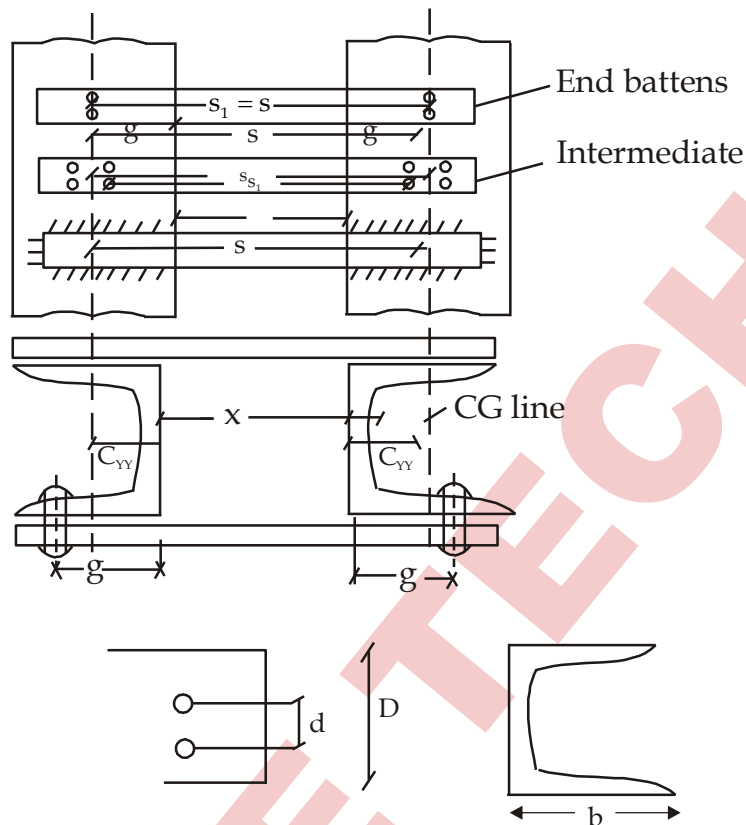
$$F_2 = -V/2 \sin \theta \quad (\text{Compression})$$

### Forces in rivets:-



$$\text{Forces in rivet} = 2F \cos \theta$$

## Design of battens:-



$C$  = Length of column component between any two battens.

$X$  = Clear spacing between two battens

$a = x + 2C_{yy}$  = distance between the CG of two column components.

$S_1$  = distance between two inner rivets or bolts.

$S$  = CG of distance between C.G. of rivet group or bolt group or weld.

$S = X + 2g$

1. In case of battens the effective length of column is increased by 10%
2. Minimum number of battens should be 4.
3. To avoid local buckling between any two battens slenderness ratio should not be greater than

$$\lambda = \frac{C}{\gamma_{\min \text{ component}}} \neq 50 \neq 0.7\lambda \text{ whole}$$

1. Effective depth of intermediate batten:-

$$d \neq 3/4 a$$

$$d \neq 2b$$

$a$  = distance between CG of two column component

$b$  = width of flange of column component.



2. Effective depth of end battens:-

$$d \neq a$$

$$d \neq 2b$$

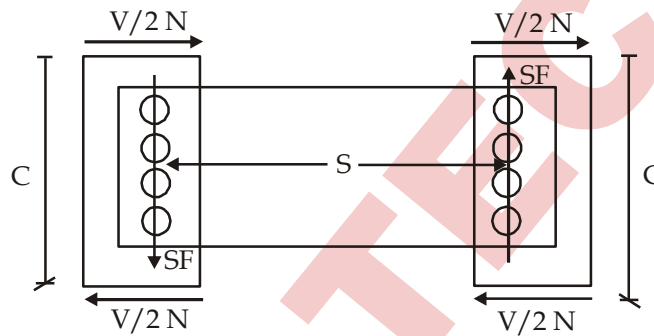
3. Thickness of batten:-

$$t = \frac{S_1}{50}$$

$S_1$  = distance between two inner rivets or bolts.

4. Forces in battens:-

Design for - 2.5% of column axial load

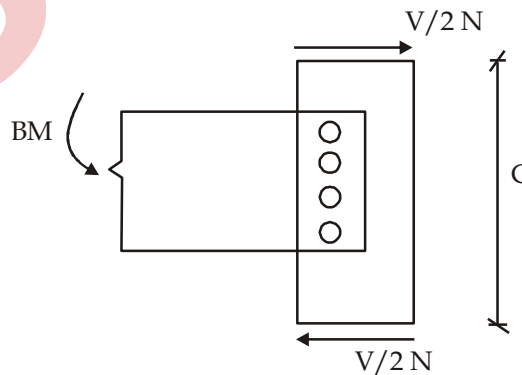


If  $N$  = Number of batten plane

$$SF \times S = \frac{V}{2N} \times C + \frac{V}{2N} \times C$$

$$SF = \frac{VC}{NS}$$

5. Bending moment



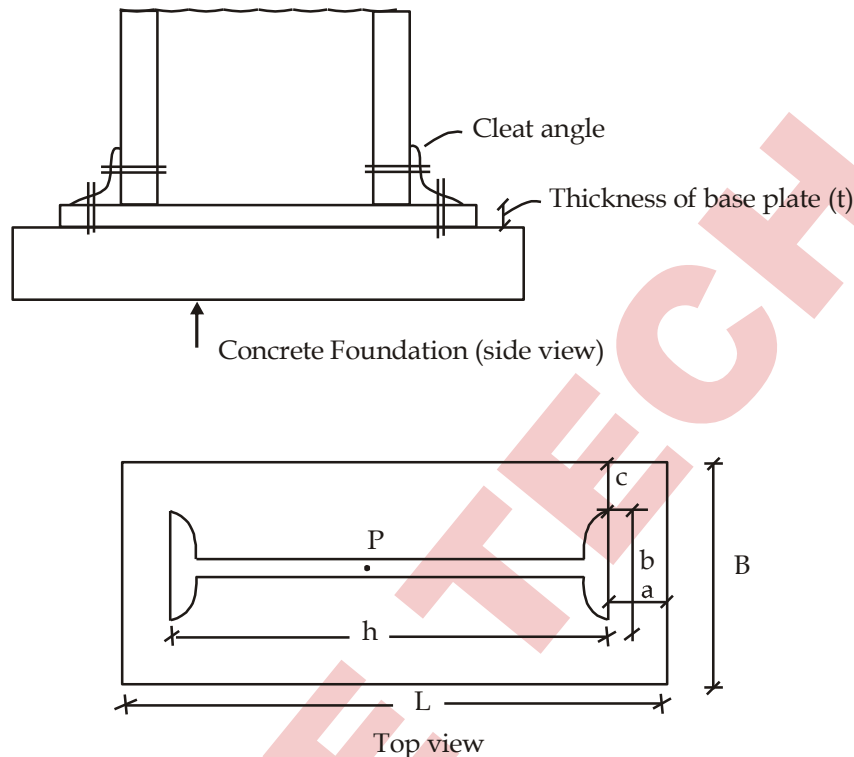
$$\sum M_z = 0$$

$$BM = \frac{VC}{2N}$$

## Design of base plates:-

### Types of base plates:-

#### (a) Slab base:-



$$a \geq c$$

If ends of column is properly maintained then it is assumed that the bearing will develop between the column and bearing plate.

- It is assumed that 50% of load is passing through cleat angle. (machine end)
- If ends are not properly machined then cleat angles will be design for total load which is to be transfer to the foundation.

WSM:-

$$t = \sqrt{\frac{3w}{b_s} \left[ a^2 - \frac{c^2}{4} \right]}$$

$w$  = Pressure on the base plate

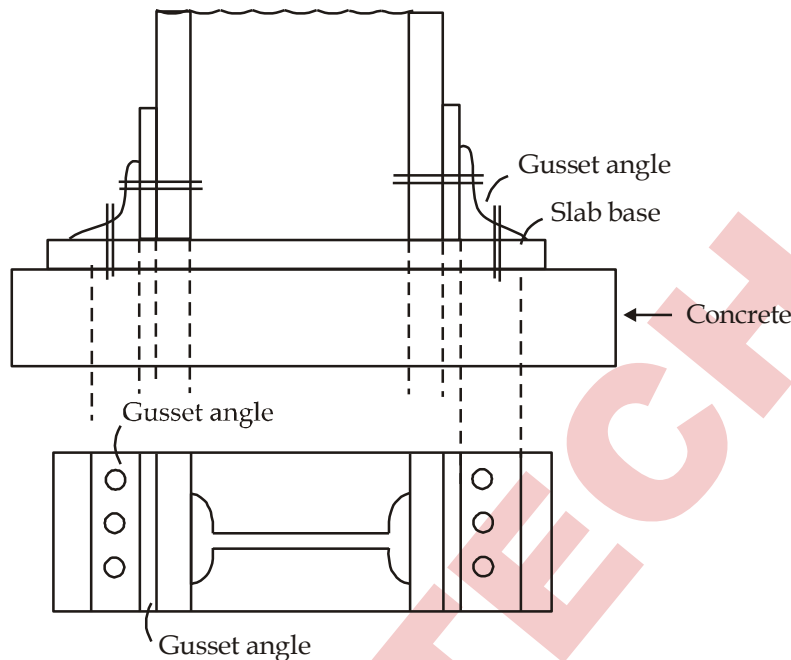
$$b_s = 185 \text{ N/mm}^2$$

LSM:-

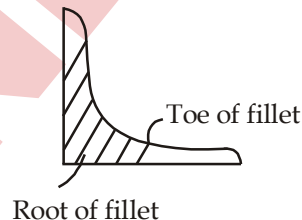
$$t = \sqrt{\frac{2.5W_u}{f_y / 1.10} \times (a^2 - 0.3c^2)}$$

$W_u$  = Factored pressure on base plate ( $w \times 1.5$ )

## (b) Gusseted slab base:-



- In a gusset slab base, base plate, gusset plate, gusset angle are used.
- Gusset plate and gusset angle are used to increase the stiffness of base plate against bending.
- The critical section for B.M. is taken at toe of fillet.

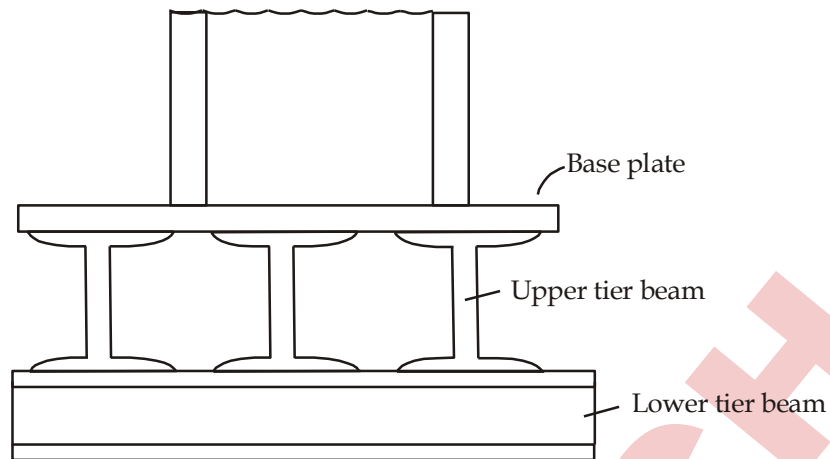
3. **Grillage Foundation:-**

Requirements of Grillage foundation:-

1. The load on the column are extremely heavy base plate requirement in such case will be of very large dimension.
2. Bearing capacity of the soil on which concrete foundation is to be placed may be poor which results in large dimension of concrete foundation and base plate.

**Assumptions for grillage foundation:-**

1. Load from the column is uniformly distributed over the base plate.
2. Each tier of beam distribute the load uniformly to lower tier.
3. Intensity of bearing pressure is uniform.

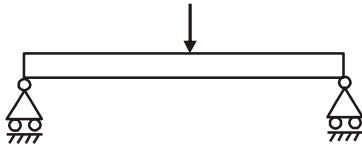


- Minimum spacing between flange of two consecutive beams is not less than 75 mm.
- A concrete cover of minimum 100 mm is provided all around the beam arrangement.

# CHAPTER - 7

## (DESIGN OF FLEXURAL MEMBER)

1.



Flexural member:- member subjected to transverse load.

Flexural Formula:-

$$\frac{M}{I} = \frac{f}{y} = \frac{E}{R}$$

$$f_{\max} = \frac{M}{I} \cdot y_{\max}$$

$$\frac{I}{y_{\max}} = Z = \text{sectional modulus} \quad \text{where } Z = \text{bending strength of section}$$

$$f_{\max} = \frac{M}{I/y_{\max}}$$

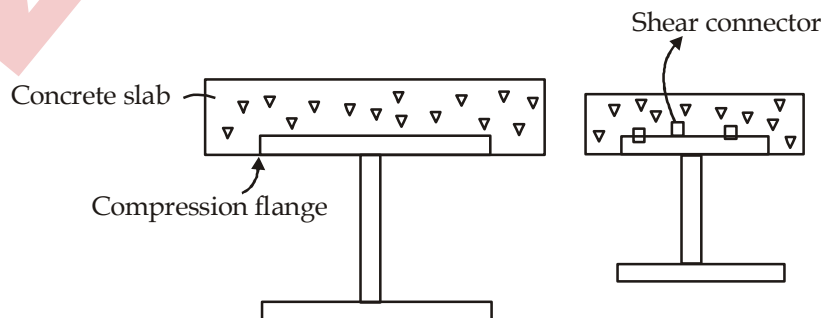
$$f_{\max} = \frac{M}{Z}$$

WSM:-

$$f_{\max} = 0.66 f_y \text{ for compression and tension.}$$

(a) **Laterally restrained beam or laterally supported beam:-**

If compression flange of beam is completely restrained against lateral movement then it is called laterally supported beam.

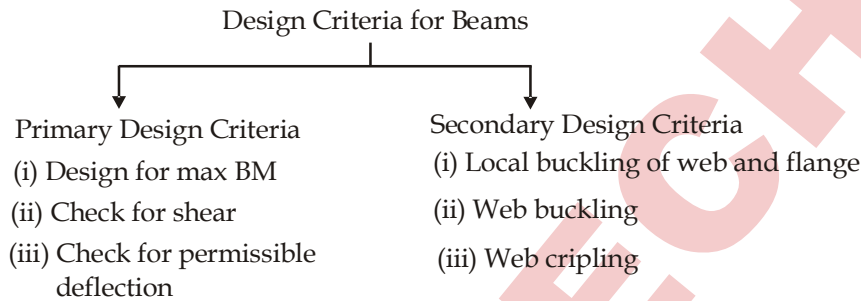


Permissible bending stress in compression flange =  $0.66F_y$

- (b) **Laterally unsupported beam:-** If compression flange of beam is free to move laterally then this type of beam is called laterally unsupported beam.

$$\sigma_{ac} = \min \left\{ \begin{array}{l} f_{cc} \times 0.66 f_y \\ \left[ (f_{cc})^n + (f_y)^n \right]^{1/n} \\ 0.66 f_y \end{array} \right.$$

**Note:-** In the beam design it is assumed that bending moment is resisted by flange and shear force is resisted by web.



### 1. Primary design criteria:-

Design steps for beam:-

(Laterally supported beam)

Step I:- After determination of maximum B.M.

$$Z_{req} = \frac{\text{Max BM}}{\sigma_{bt} \text{ OR } \sigma_{bc}}$$

Step II:- Provide suitable section according to  $Z_{req}$  by the use of steel table.

Step III:- Check for shear

$$\tau_{v \text{ calculated}} = \frac{\text{Max SF}}{d_w \times t_w} \neq 0.4 f_y$$

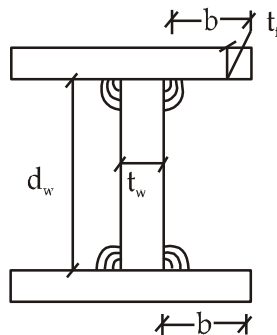
Step IV:- Check for maximum deflection

$$\delta_{Max} \neq \text{span}/325$$

$$\frac{5}{384} \cdot \frac{wl^4}{EI} \neq \text{span}/325$$

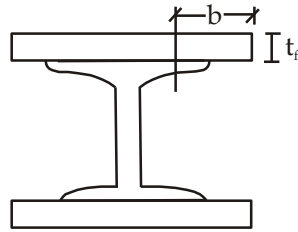
### 2. Secondary design criteria:-

- (i) **Local buckling of web and flange:-**



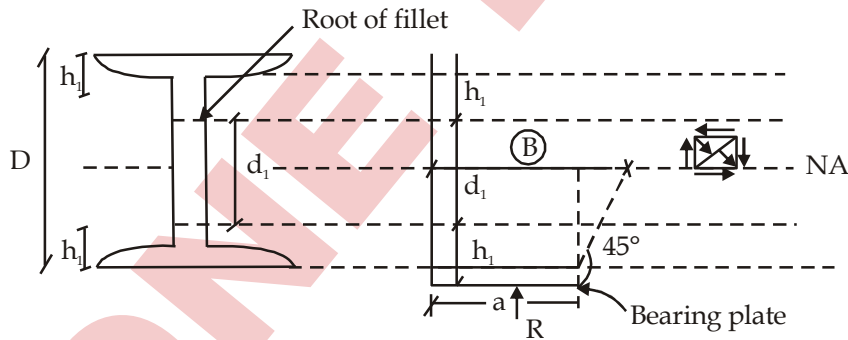
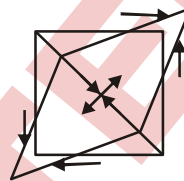
$$b/t_f \neq 16$$

$$d_w/t_w \neq 50$$



$$b/t_f \neq 50$$

2. **Web buckling:-** Web buckling will occur due to diagonal compression which is produced due to high shear stresses.



$$B = (a + h_1 + d/2)$$

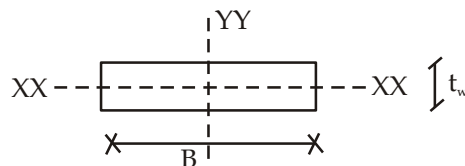
$$B = (a + D/2)$$

Length of column =  $d_1$

Effective length of column =  $d_{1/2}$  (both end fixed)

Slenderness ratio of  $d_1$  length column

$$\lambda = \frac{l_{eff}}{\gamma_{min}}$$





$$I_{\min} = I_{XX} = \frac{B \cdot t_w^3}{12}$$

$$\text{Area} = B \cdot t_w$$

$$\gamma_{\min} = \sqrt{\frac{B \cdot t_w^3}{12 / B t_w}} = \frac{t_w}{12} = \frac{t_w}{2\sqrt{3}}$$

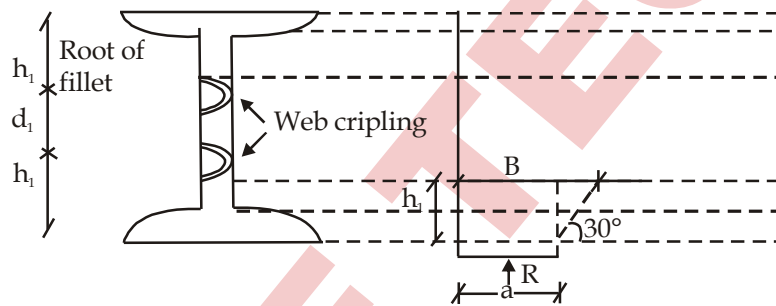
$$\left[ \lambda = \frac{L_{\text{eff}}}{\gamma_{\min}} = \frac{d_{1/2}}{t_w / 2\sqrt{3}} = \frac{\sqrt{3} d_1}{t_w} \right]$$

By the use of  $\lambda$ ,  $\sigma_{ac}$  will find out

Buckling strength of column

$$= B \cdot t_w \times \sigma_{ac} \geq R \text{ prevent for web buckling}$$

**Web crippling or web crippling:-**



$$B = (a + \sqrt{3} h_1)$$

Bearing area

$$= B \times t_w$$

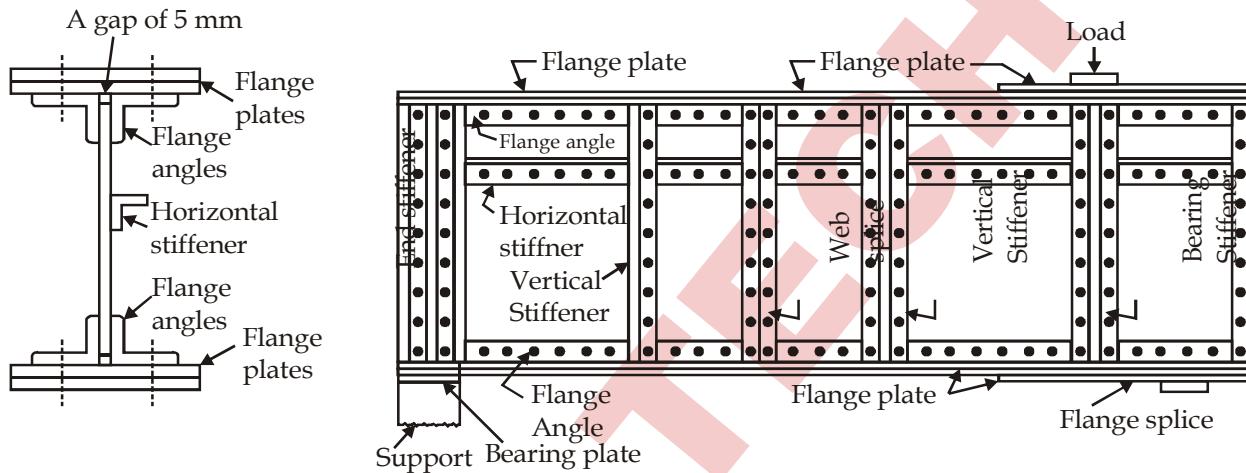
**Bearing stress at root of fillet:-**

$$= \frac{R}{B \cdot t_w} \leq 0.75 f_y$$

↑  
To prevent web crippling  
 $0.75 f_y$  permissible bearing stress

# CHAPTER - 8

## (PLATE GIRDER)



### Some Important point:

- 5mm is provided to isolate flanges from web so that direct bearing action on web avoided and web should exist shear only.
- Effective flange area in compression zone:-

$$A_f + \frac{A_w}{6}$$

$A_f$  = area of compression flange

$A_w$  = area fo web

- Effective area in tension zone will be

$$A_f + \frac{A_w}{8}$$

- Area of flange angles will be equal

$$\frac{A_f}{3}$$

- If unequal angle sections are used then longer leg will be connected to flange (To reduce bearing)

- Self wt. of plate girder  $\frac{W}{300}$

where

W = Total live load or imposed load

- Economical depth of plate girder

$$d_e = 1.1 \sqrt{\frac{M}{\sigma_{bc} \times t_w}}$$

where

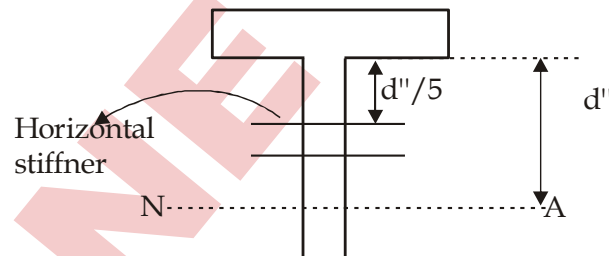
$\sigma_{bc}$  = Permissible stress in bending compression

$$= .66 f_y \text{ or } \frac{0.66 f_y \cdot f_r}{\left[ (f_{cr})^n + (f_y)^n \right]^{1/n}}$$

- ☞ **Vertical stiffeners:** These stiffeners are provide to resist chances of web buckling due to shear.
- ☞ **Horizontal stiffeners:** These are provided in compression zone to reduces chances of lateral buckling of web due to bending compression. These are also called as longitudinal stiffener.
- ☞ **Load Bearing stiffeners:** Load bearing stiffeners are used to avoid crippling by transferring load from are flange to another flange.

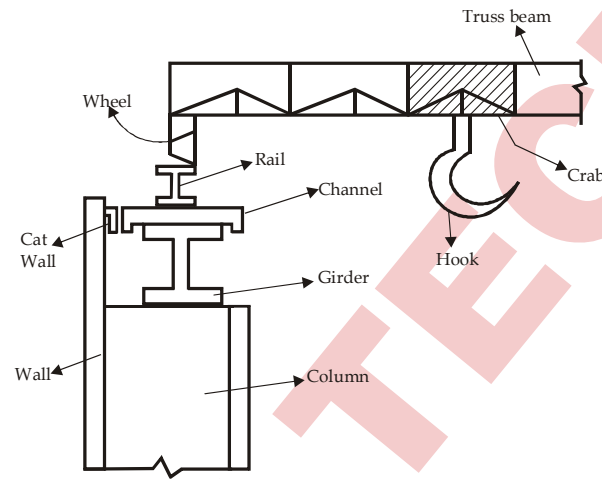
## Criteria for providing stiffeners

	WSM	LSM
1.	$\frac{d_1}{t_w} < 85$ No stiffner required	$\frac{d_1}{t_w} < 90$ No stiffner required
2.	$85 < \frac{d_1}{t_w} < 200$ Only vertical stiffners required	$90 < \frac{d_1}{t_w} \leq 200$ Only vertical stiffner required
3.	$200 < \frac{d_1}{t_w} \leq 250$ Along with vertical stiffners 1 horizontal stiffner is provided 1/5 of distance of compression flange from N. A	$200 < \frac{d_1}{t_w} \leq 250$ Along with vertical stiffners 1 horizontal stiffner is required 1/5 of distance of compression flange from N. A
4.	$250 < \frac{d_1}{t_w} \leq 400$ → Vertical stiffners +2 horizontal stiffners → Second horizontal stiffner will be provided at N.A.	$250 < \frac{d_1}{t_w} \leq 400$ → Vertical stiffners +2 horizontal stiffners → Second horizontal stiffner will be provided at N.A.



# CHAPTER - 9

## (GANTRY GIRDER)



☞ Gantry girder is subjected to these perpendicular forces:

- **Vertical Force:** Due to self wt, wt. of material. wt of crab, truss girder.
- **Lateral load:** Due to different positions of crab.
- **Longitudinal load:** Impact due to movement of wheel in longitudinal direction.
- At a time where wheel will move crab will be at stop, and when crab will move wheel will be at stop means it is assumed that lateral and longitudinal forces will not act together.

☞ **Load Combinations:**

- (i) Vertical + lateral
- (ii) Vertical + longitudinal

- **Rail Channel:** Rail channel will increase the lateral strength of gantry girder.
- **Cat wall:** Cat wall is used to resist lateral movement of compression flange so that the chance lateral buckling can reduce significantly.

- **Channel section:** Channel section over gantry girder is not used to provide torsional restraint because it is not connected with web.
- **Web stiffeners:** Are used to increase torsional resistance of web.
- The following impact factors are used for moving loads:
  - **25%** of standing load .(For electrically operated)
  - **10%** of standing load (For hand operated)
 {These two factors will multiplied in vertical load}
- **Longitudinal load:** 5% of static load.
- Vertical deflection of gantry girder should not exceed value as:

1. When the crans are manually operated. deflection should not exceed =  $\frac{L}{500}$

where

L = span of gantry girder.

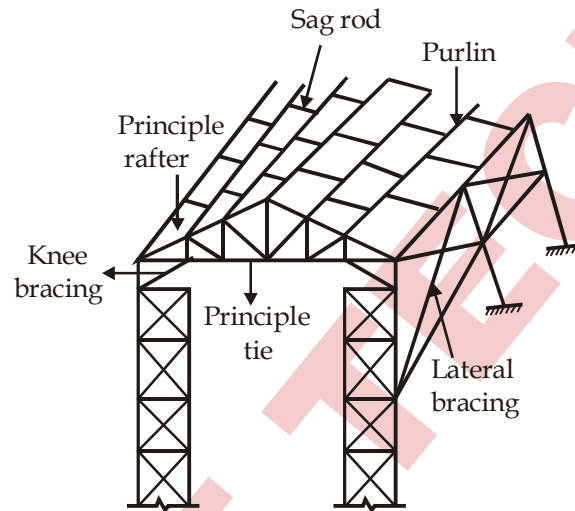
2. Electric overhead travelling crans upto 500 kN capacity :

$$\text{Deflection} = \frac{L}{750}$$

3. Electric overhead travelling crans for greater than 500 kN capacity,  $\Delta = \frac{L}{1000}$

# CHAPTER - 10

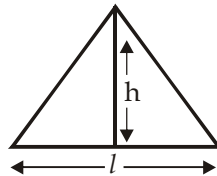
## (ROOF TRUSSES)



### Pitch of Roof:

$$P = \frac{h}{L}$$

$$\tan \theta = \frac{2h}{L}$$



1. Always consider wind load perpendicular to the roof.

Where

$L$  = span of truss

2. Economical spacing of truss varies from  $\frac{L}{3}$  to  $\frac{L}{5}$
3. Reversal of stress takes place due to wind or earthquake.



4. Bracings are designed to resist lateral loads.
5. Sag rods are designed for tensile load.
6. Purlins are designed as continuous member over trusses. (Continuous beam)
7. **Design of purlins:**

- Roofing material is supported by purlins

- The depth of purlin  $\leq \frac{L}{45}$

L = Length of purlin

- Width of purlin  $\leq \frac{L}{60}$

- Maximum B.M. in purlin  $\leq \frac{WL}{10}$

W = Total load from roof.

- After calculating maximum B.M. check for bending stress generate in purlins.

$(\sigma_{bc})$  calculate  $\leq 0.66 f_y$

$(\sigma_{bt})$  calculate  $\leq 0.66 f_y$

- **Check for deflection:** Deflection  $\leq \frac{L}{325}$

#### ☞ Calculation of load:

- **Dead Load :** For a given spacing of trusses as 4 m and pitch equal to 1/4, the self wt. of truss is taken as:

$$W = (l \times B) \left\{ \frac{l}{3} + b \right\} \text{ kg}$$

l = length of span

B = spacing of truss

☞ **Live load calculation:**

(i) When  $\theta < 10^\circ$  : L.L. =  $0.75 \text{ kN/m}^2$

(ii) For  $\theta > 10^\circ$  L.L. =  $0.75 - (\theta - 10) \times 0.02$

$$\nless 0.4 \text{ kN/m}^2$$

- Maximum roof angle  $\nless 30^\circ$

☞ **Snow load :-** For every 1 mm thickness of snow.

$$\text{Snow load} = 2.5 \text{ N/m}^2$$

- To minimize the total cost of roof truss the ratio of cost of truss to cost of purlins should be 2.

☞ **Wind load:** When wind blows it generates wind drag (pressure) on roof surface.

- The wind pressure depends on slope of roof.

- Design wind pressure

$$\begin{aligned} P_z &= 0.6V_z^2 \\ &= K \cdot V_z^2 \end{aligned}$$

where

$P_z$  = design wind pressure.

- $V_z$  = design wind velocity

$$V_z = V_b \cdot K_1 \cdot K_2 \cdot K_3$$

$V_b$  = Basic wind speed at height 10 m

$K_1$  = Probability factor (Risk factor)

$K_2$  = Terrain, height of structure and size factor  $K_3$  = Topography factor

# CHAPTER - 11

## (MISCELLANEOUS TOPICS)

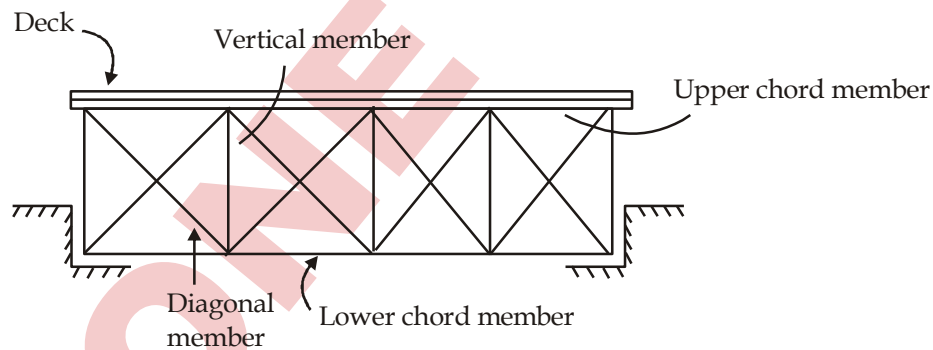
### Steel Bridges

Types of steel bridges:-

1. T-section bridge
2. Plate girder
3. Truss girder bridge
4. Suspension bridge
5. Cable stayed bridge

### Truss girder bridge:-

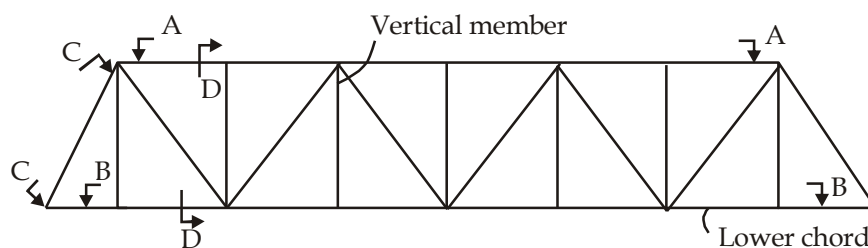
1. Deck type bridge (movement of vehicle at upper chord)
  2. Through type bridge (movement of vehicle at lower chord)
1. **Deck type bridge:-**



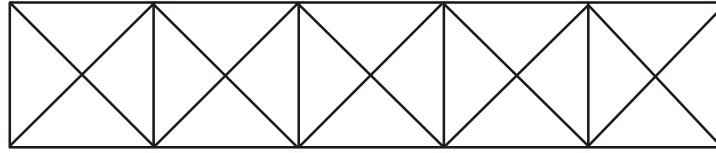
Front view

- In this type of bridge floor system is supported on top chord members.
- In this type of bridge sway bracings are provided to prevent rectangular shape of bridge.

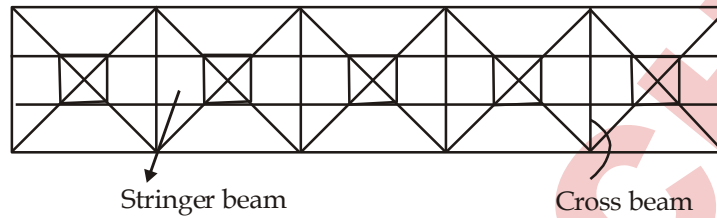
2. **Through type bridge:-**



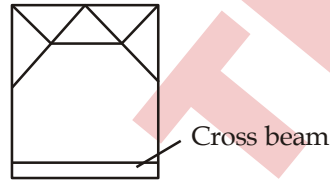
(a) Front view or Elevation of truss girder



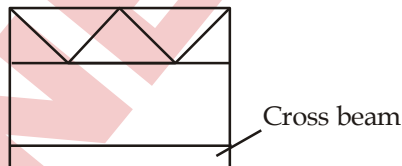
(b) Top lateral bracing view (A - A)



(c) Bottom lateral bracing view (B - B)



Portal bracing or end bracing view at C - C



Sway bracing view at D - D

This bridge consist of 1 main vertical trusses.

1. Floor system with bottom lateral bracings
  2. Top lateral bracing
  3. Portal bracings or end bracings
  4. Sway bracings
- Top chord bracings and bottom chord bracings are provided to resist lateral load due to wind earthquake and raking forces.
  - Top chord bracings are designed for a load addition to wind load = 2.5% of sum of compression forces in top chord.
  - Bottom chord bracings are subjected to wind load raking forces and longitudinal forces due to movement of vehicles.

- Portal bracings and sway bracings are provided to prevent rectangular shape of bridge.
- Portal bracings are designed for wind load and 1.25% of load on both top and bottom chord members.
- Sway bracing are provided to transfer 50% of top panel wind load to bottom panel.

Plate girder, Gantry girder, Roof truss

### STEEL TANKS

- These are built to contain water, petrolium, other chemicals.

#### Types of steel tanks:-

1. Rectangular type steel tank
2. Circular type steel tank
3. Pressed type steel tank

#### Pressed type steel tank:-

- Made of plates which are pressed in hot condition to get connected.
- These type of tanks have very less chances of seepage.
- The minimum capacity of pressed tank as per IS : 805 is 1950 litres.
- In case of steel tanks when rivets are provided there spacing in maximum case should not be greater than  $10t$ . and minimum spacing should not be less than  $3d$ .
- The minimum thickness of plates used for steel tanks should not be less than 6 mm in normal conditions. If storage of salt is taking place, minimum thickness should not be less than 5 mm.
- These two criterias are not applicable for roof of tank.

#### Permissible stresses in steel tanks:-

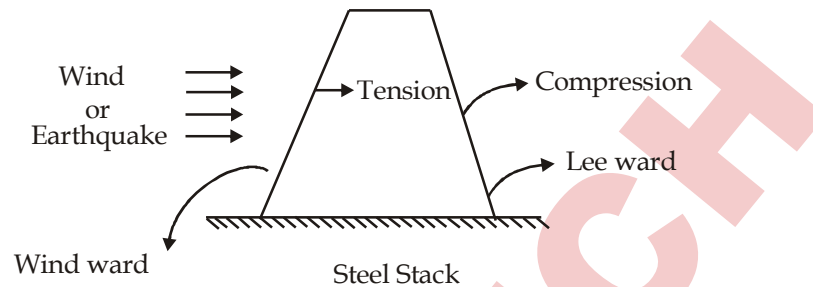
- The permissible stresses are reduced to take care of corrosion.

#### For steel plates (Permissible stresses):-

1. Direct Tension/Compression =  $0.8 \times 0.6 f_y$
  2. Bending Tension/Bending compression =  $0.8 \times 0.66 f_y$
  3. In all other conditions =  $0.8 \times$  working permissible stresses.
- Efficiency of joing (connection) is considered as 70 to 75% in case of tension.

### STEEL STACKS (CHIMNEY)

- Stacks are designed as per IS : 6533.
- In stacks minimum thickness should not be less than 6 mm except upper portions where thickness should not be less than 8 mm (To resist more corrosion at top)
- Width of steel plate is taken as 0.9 to 2.5 m.



- Temperature in steel stacks  $\neq 400^{\circ}\text{C}$
- Permissible stresses are considered as 0.7 to 0.8 times of normal/working permissible stresses.
- In case of wind and earthquake, wind ward side is designed for tension and lee ward side is designed for compression and permissible stresses are increased by 15% in this case.